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TC-1002740

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HANGAR NOISE

Moving ahead with WSDRS (CAR 591)

Transport Canada has taken the initiative of moving to a paperless work environment over the past five years.

In order to facilitate a paperless environment, Continuing Airworthiness has developed and promoted the use of a web-based reporting system that allows the Canadian Aviation Document holders to submit safety defect information online.

The first version of the web service difficulty report system (WSDRS) was released in 2001. Over the past six years Transport Canada (TC) and industry have learned many things using online SDR reporting. WSDRS is an easy and effective means of reporting safety defect information as required by *Canadian Aviation Regulation* (CAR) 591.

Record keeping for users is simplified as WSDRS electronically stores all submitted records; communication with your principal maintenance inspector (PMI) and TC headquarters (HQ) has improved; everyone is kept in the loop from the initial reporting by the submitter, to the development of corrective action; and the reduction of paper which in turn is good for the environment. The advantages of using WSDRS are obvious, and presently 90% of Canadian SDR's are submitted through WSDRS.

TC now has more than 1.2 million records, which consists of Australian, USA, and Canadian data in the WSDRS database. The need was identified to redevelop

Feedback is published quarterly by the Continuing Airworthiness Division of Transport Canada, informing the aviation community of reported day-to-day problems that affect aircraft airworthiness in Canada.

Reprints of original Feedback material are encouraged, but credit must be given to Transport Canada's Feedback magazine. Please forward one copy of the reprinted article to the Editor.

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> Sylvie Barrick, Editor Feedback Transport Canada (AARDG) Place de Ville, Tower C Ottawa ON K1A 0N8

E-mail: *barrics@tc.gc.ca* Tel.: 613-952-4360 Fax: 613-996-9178 Internet: *http://www.tc.gc.ca/cawis-swimn/* WSDRS, enabling the application to more easily extract information required to make safety-related decisions.

We have enhanced WSDRS and provided improved search capabilities for registered users, developed a new users inbox and made the application more user-friendly. In the new application, your existing username and password is retained.

 + 	Transport Canada	Transports Canada			Canada
R	Français Home Air	Contact Us About us Marine	Help Media room Rail	Search Environment Road	Canada Site Emergencies Major issues
RANSPORT		ote that this applicat s due to scheduled			
WSDRS	apologize This site i	for any inconvenien	ice this may caus	e.	
Registered Users Jsername / Password	Microsoft	pher. Ip Blocker feature of page " <u>Block Pop-ur</u> ns regarding the use	Windows with Ir	ternet Explorer	contains
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Help SDR Fax/Paper Form Logic Chart	of Canadia Country of	s mainly for use by o an registered aeron: Type Design respo the site are welcom	autical products o nsibility.	r products for wh	ich Canada is the
Related Links Continuing Airworthiness Web Information System	 Su Qu Trating 	ered users can utili bmit SDRs lery the SDR datab ick and store subm date previously sub	ase iitted SDRs		

If you are not registered, you should be! Although you will be able to submit safety defect information by other methods acceptable to the minister, we strongly encourage you to submit them using WSDRS.

The articles contained in **Feedback** are derived from *Service Difficulty Reports* (SDRs) submitted by Aircraft Maintenance Engineers (AMEs), owners, operators and other sources in accordance with *Civil Aviation Regulation* (CAR) 591.

Service Difficulty Reports (SDR) are normally published verbatim. Transport Canada assumes no responsibility for the accuracy or content of any of these reports. Only grammatical or spelling errors are corrected and content may be reduced as well as personal references deleted.

All defects or occurrences should be reported to Transport Canada through the Service Difficulty Reporting Program. For additional information about this program or concerning an article in feedback magazine, contact your nearest Transport Canada Centre.

Feedback est aussi disponible en français.

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TP 6980E

Users who only need to query the SDR database do not need to be registered. For all non-registered users, we are expanding the basic search and result export capabilities. This new look is similar to the Airworthiness Directive website as both systems come under the same CAWIS umbrella. *****

FIXED WING

BEECH 99

SDR# 20070212007

Elevator Torque Tube Severely Cracked

The tail cone was removed to carry out *service bulletin (SB) 2145*, elevator torque tube fitting inspection, which is due every 100 hours. The technician then noticed that the elevator torque tube itself, although not part of SB 2145, was severely cracked. The last detailed inspection of the torque tube was some 245 hours ago.

The crack originated from the horn attachment taper pin and spiralled approximately 270 degrees around the circumference of the actual torque tube.

This general area is inspected every 100 hours, thus it appears that this crack developed rapidly within this timeframe.



Fortunately, the crack was discovered thus preventing a potentially serious accident and loss of elevator control authority. The type certificate holder has been notified.

The SDR database contains several previous reports related to torque tube cracks originating at the taper pinhole, but none as severe as this event. \mathbf{x}

BEECH 1900D

SDR# 20070212010

Main Wheel Assembly Missing

After landing and all of the passengers were unloaded for refuelling of the aircraft, the pilot walked around the aircraft to speak with the fuel truck driver. It was then noticed that the right main gear was missing the complete tire and wheel assembly. Maintenance was immediately contacted and the aircraft was grounded.

Further inspection of the axle stub revealed that the inner races of the bearing, nut and safety hardware were still attached. This aircraft had recently been serviced at a foreign manufacturer's repair facility prior to purchase by this Canadian operator.

As a precautionary measure, the remaining three (3) main wheel assemblies were removed and inspected for condition and security. Two of these remaining wheel assemblies were found to have been grease-coated on the outside of the bearings, but were not "grease-packed" inside (as per manufacturer's instructions). Subsequently, the wheel bearings were cleaned, inspected and properly repacked with the specified lubricant.

The missing main wheel assembly was located in the area near the active runway.

The domestic operator received excellent follow-up support from the type certificate holder who inspected the complete aircraft and completed the phase inspection.



Transport Canada Civil Aviation (TCCA) does have previous service history on this wheel-bearing problem. Subsequently, TCCA issued Service Difficulty Alert (AL 2006-02) to inform operators and maintenance personnel that the present wheel-bearing lubrication schedule may be inadequate. **X**

BEECH A100

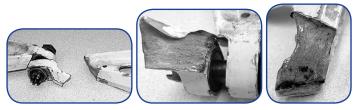
SDR# 20071024010

Landing Gear Torque Link Failed

After touchdown and decelerating through 60 knots indicated airspeed (KIAS), the aircraft began to violently shake, shudder and appear to skip while veering toward the left side of the runway. Control of the aircraft was maintained as it was brought to a slow taxi at which time the conditions ceased.

Upon examination of the aircraft, it was discovered that the left main gear lower torque link, part number (P/N) 50810323, had failed. The torque knee was replaced and the aircraft was returned to service.

The operator initiated a fleet-wide aircraft non-destructive testing (NDT) inspection that revealed three additional cracked torque links. The latest revision of Raytheon Beech safety letter (SL) 0516-200 incorporates torque



link hardware and bushing replacement, however there is no further requirement to conduct a NDT inspection as recommended in the original SL 0516-200.

Transport Canada recommends operators to inspect the main landing gear torque link assemblies for defects. NDT may need to be reconsidered.

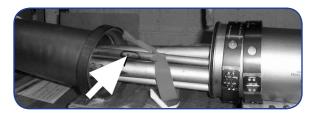
The torque link has been sent to the Transport Safety Board (TSB) lab for analysis and the type certificate holder has been advised of this failure. \clubsuit

BOMBARDIER CL600 2B19 (RJ100)

SDR# 20071005001

Fuel Line Chafed

During a heavy maintenance check the fuel shroud assembly, fuel lines at frame 559 were removed to facilitate a sheet metal repair.



It was discovered that both shroud and fuel lines had chafing damage beyond acceptable limits. This same repair was carried out on other aircraft in the fleet and similar damage was found in the same location.

The fuel lines and shroud assembly have been ordered.

If you are in this area, inspect the lines carefully. 🛠

BOMBARDIER CL600 2B19 (RJ100)

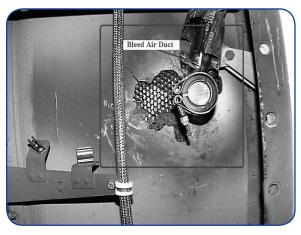
SDR # 20070506002

Engine Inlet Cowl Bleed Air Damaged

After the engine fan cowl was removed during a scheduled maintenance check, structural damage was observed on the inlet cowl. It was apparent that the capped 14th stage bleed air duct had leaked hot bleed air and caused heat damage to the immediate area.

It was later confirmed that an incorrect clamp had been used.

Unfortunately, human factor errors still occur and directly led to this event. Always take the time to verify that the correct manufacturer's illustrated parts catalogue (IPC) part number is installed. 🛠



BOMBARDIER CL600 2B19 (RJ100) SDR # 20070404011

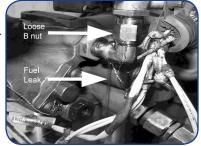
Auxiliary Power Unit (APU) Ground Fire

The technician was troubleshooting the APU, GTCP36-150RJ, following an "auto-shutdown" event due to high oil temperature indications. Ground inspection revealed that the electrical connector for the oil temperature sensor was found partially installed. The connector was secured and the APU enclosure inspected with no evidence of fuel or oil leakage.

Shortly after APU functional start and with both air condition packs operating normally, a strong smell of fuel was noted in the galley and cockpit area. Fuel was also seen coming from the APU drains and the APU was immediately shut down.

A technician climbed into the aft equipment bay and began to open the APU oil-servicing panel, when a fire flared up. The cockpit crew employed the onboard extinguishing agent and the ground crew also utilized a dry chemical agent into the APU enclosure.

The APU was removed and sent to the original equipment manufacturer (OEM) for further examination. There was no evidence of fire damage other than some fire suppressant contaminants.



When the APU was hooked up to the test cell, a stream of fuel was seen coming from the elbow fitting between the fuel control and the fuel shut-off valve. Closer examination found that the fuel leak originated at the B-nut fitting. The B-nut was tightened approximately ³/₄ of a turn and the fuel leakage stopped. Other operational checks were carried out with no defects found.

Transport Canada recommends always double-checking fluid carrying lines following assembly. 🛠

CHAMPION 7ECA

SDR# 20060928003

Elevator Travel Impeded due to Failed Gusset

After recovering from an aerobatic manoeuvre, the pilot noted that the "full down" elevator travel could not be obtained. An uneventful landing was carried out.



Maintenance personnel removed the elevator bell crank cover and found that the left gusset from the aft stringer support had fallen into the belly of the aircraft. Over a period of time, the gusset worked its way back and became lodged in the narrow clearances around the elevator bell crank.

Examination of the gusset, support and rivets indicated that the airframe fabric vibrations occurring during flight had loosened the soft aluminium pop rivets and allowed the gusset to become free.

Fortunately, the dislodged gusset was found before a complete elevator jam occurred. It is unknown how long the gusset was in this area. \clubsuit

DE HAVILLAND DHC 8-100

SDR# 20060831002

Nose Landing Gear Shock Strut Housing Cracked

During a scheduled maintenance check, task card 3220/01 & 02, visual inspection & lubrication of nose landing gear (NLG) assembly, a crack was discovered in

the NLG outer cylinder, part number 8814-7. The nose landing gear assembly was immediately replaced.

The assembly was forwarded to Messier-Dowty for investigation, which revealed evidence



of earlier rework. The crack origin showed evidence of re-work resulting in a non-conforming surface finish and a visible absence of shot peening. Bombardier/ de Havilland has made reference to two earlier investigations on outer cylinder separations. Bombardier/ de Havilland subsequently initiated an information campaign informing fleet operators on the non-recommended practice of single engine taxiing, which produces stress on the NLG assembly.

Although the origin of this crack was due to nonconforming practices, the single engine taxiing appears to have been a contributing factor. x

DE HAVILLAND DHC 8-300

SDR# 20061205001

Main Landing Gear Yoke Fitting Cracked

During a scheduled A-check the right main landing gear yoke was found to have a cracked fitting at the auxiliary actuator attachment point.

The landing gear



manufacturer Messier Dowty, and the aircraft type certificate holder, Bombardier, are conducting an investigation into this yoke fitting fracture.

Transport Canada wishes to disseminate this information to the operators of this aircraft type even though an investigation is on-going. An alert technician discovered this discrepancy before an incident or accident occurred. *****

DE HAVILLAND DHC 8 301

SDR# 20070510007

Nose Landing Gear Uplock Actuator Assembly Fitting Cracked

During approach for landing, the pilot reported that the nose landing gear (NLG) would not extend to the "down and locked" position. The aircraft carried out a fly-pass over the airport control tower that stated that the nose gear was extended but did not appear to be locked. The pilot cycled the landing gear several times and then finally got the cockpit indication that the landing gear was "down and locked". An uneventful landing was carried out.

The aircraft was ferried with the landing gear secured in the down position to a maintenance base for further inspection. Maintenance personnel then discovered that the NLG uplock actuator fitting assembly was cracked at the actuator attachment point. The discoloration in the crack area indicated that the crack propagation occurred over a period of time. The SDR submitter recommended that a more focused specific visual or NDT inspection to be carried out on this fitting. Additionally, the NLG trunnion sidewall web and top web were also found cracked. The



"Z" member, P/N 85311420, on the right side of the wall was also found cracked. Repairs were accomplished in accordance with the manufacturer's instructions.

The submitter comments that as aircraft accumulate lengthy time in service, the need for extra attention during routine inspections is recommended. x

DE HAVILLAND DHC 8 311

SDR# 20070427008

Aileron Cable Misrouted



The pilot reported that roll control inputs were "stiff", especially left to right applications. Ground functional tests could not duplicate this snag and nothing seemed adverse with the roll disconnect.

Maintenance inspection found that the left outboard aileron cable was routed around the wrong side of the cable guard. The aileron cable was inspected, routed correctly and re-rigged, and an Independent Check carried out.

The cable guard had to be replaced due to "wear" and the aircraft was returned to service.

Fortunately, it was the cable guard that was worn and not the actual aileron cable. An Independent Check is a mandatory safety measure that is required if engine or flight control rigging is disturbed. 🛠

EMBRAER ERJ 190 100 IGW

SDR# 20071101003

Dragging Our... Wheel

The aircraft landed normally, taxied to the gate and stopped short of its final position. The crew reported 60% N1 required (single engine taxi) to move the aircraft from the hold short to the gate. The ground crew arriving to the aircraft heard a noise from the right gear as the aircraft taxied to the gate. The pilots reported that there were no fault indications displayed in the cockpit. Upon inspection of the right landing gear, it was identified that the No. 4 wheel was dragging and interfering with the brake carrier.

Further inspection found the above wheel bearing seized on the gear axle, with the No. 4 tire and the brake damaged. Discoloration of the axle was noted due to excessive heat build up.

The right landing gear assembly, brake and tire were replaced as per the *Aircraft Maintenance Manual* (AMM).

The investigation continues to identify the root cause of the problem and component strip shop reports are requested.



Transport Canada (TC) will update this issue when the root cause is determined. \boldsymbol{x}

PIPER PA 44 180

SDR# 20070815001

Downlock Spring Broken

On approach, the pilot noticed that the left (green) gear "down and locked" indicator was not illuminated. As the aircraft was slowed down on final, the "gear unsafe" light came on.

The pilots elected to go around an≠d informed flight services regarding their situation and the airport emergency response services were placed on standby. The aircraft emergency gear extension was carried out with the same result. The left main gear was extending full travel, however it was not locking down.

The pilots chose to push the emergency extension knob to its normal position and the gear retracted and extended hydraulically and a decision to land was made. The aircraft landed without incident and was followed by the fire vehicles to the hangar.

Upon investigation of the left main landing gear, maintenance discovered that the downlock spring, P/N 487495, had broken at the upper end. When functioning correctly, this spring pulls the downlock hook onto the pin and pushes the plunger to the gear indication switch.

The spring was replaced, gear retractions and extensions were carried out normally and the aircraft was returned to service.

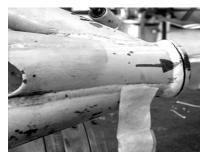
Maintainers should keep this defect in mind when inspecting this area. A thorough cleaning of the downlock may be required \mathfrak{K} .

Feedback 01/2008

ENGINES

HONEYWELL (GARRETT) TPE331-12UHR SDR # 20070507008 (FAIRCHILD SWEARINGEN SA227)

Engine Truss Mount Cracked



During a Phase 6 inspection, a crack was found on the right engine truss mount support assembly. The crack was located at the upper inboard weld joint that attaches to the engine firewall.

SDR # 20070309003

The last detailed inspection of the engine truss was previously done 300 hours ago during the Phase 4 inspection.

A supplemental inspection document 71-21-01 and a service bulletin (CC7-71-001) exist that are related to engine truss inspection.

The tubular engine mount truss assembly is mounted in four (4) places on the nacelle firewall. The truss assembly is reinforced with gussets and welded at each joint.

Inspection of the truss weld areas is critical especially following turbulent flight, hard landings and following propeller strikes. If one or more of the engine truss (or mounts) is significantly damaged, a catastrophic event could occur, including complete engine separation from the wing. \clubsuit

ROLLS ROYCE - SPEY 555-15P

SDR # 20070613001

Bypass Duct – Unapproved Repair

The engine was received from a foreign operator for inspection. During disassembly, the engine front bypass duct was found to have a non-standard patch repair that was not in accordance with the original equipment



manufacturer's (OEM) limitations. The patch repair was on the outside wall and measured 8 inches long by 2.5 inches wide.

The bypass duct was subsequently removed from service.

<u>Always</u> follow the maintenance and repair instructions published by the OEM. $\stackrel{\bigstar}{\mathbf{x}}$

PROPELLERS

MD HELICOPTERS (HUGHES) 369D

Main Rotor Blade Cracked

The pilot felt a vibration during flight and thus investigated further after landing. An 18" crack was found at Blade Station 36.

This crack was running perpendicular to the blade trailing edge. It appears that the blade became debonded because the crack followed straight from the rib towards the "C" channel before it trailed off to the side.

The SDR submitter opined that this incident might be caused by too much tension on the blades during and after blade tie-down. This may lead to "oil-canning" between the blade ribs and thus put undue pressure on the bonds.



This blade has accumulated approximately 2600 hours, which is about 2/3 of the 3500-hour life limit on this blade. FAA AD 2005-21-02 "Torque Event Inspection" applies to this particular blade. $\stackrel{\bullet}{\mathbf{X}}$

HEADS UP

LOCKHEED HS 748 2A

SDR # 20070905005

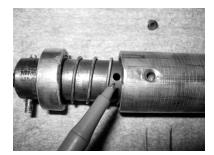
Ground Terrain Incident - Parking Brake Failed

After landing, the aircraft taxied to the ramp was then marshalled into the assigned gate. The pilot reduced engine power after coming to a full stop, set the parking brake, and waited for ground electrical power hook-ups prior to shutting down the aircraft engines. The aircraft brake pressure gauge indicated full hydraulic pressure. The aircraft was not yet chocked because the propellers were still turning.

As the pilot was looking downward and completing final shutdown checks, the aircraft began to move forward until the left propeller made contact with the ground power unit. The force of the impact severely bent and twisted the propeller blades and caused the left engine to quit running. Fortunately, ground support personnel were able to get out of the way in time and evade any of the debris that was flying around.

Maintenance troubleshooting found that three (3) rivets that secure the inner shaft to the outer sleeve of the parking brake lever assembly had failed. This failure released all of the hydraulic park brake pressure causing the aircraft to move forward.

The operator has stated that there is no maintenance visit specified for inspection



to check the condition of the rivets that secure the park brake shaft and sleeve mechanism.

Due to the propeller strike, the engine was removed and sent out for further investigation.

The above scenario is a good reminder to all ground personnel of the potential hazards of working in close proximity to an aircraft with the engine running. This is especially important around propeller-driven aircraft. \clubsuit

EQUIPMENT ADs

Transport Canada (TC) endeavours to send copies of new airworthiness directives (ADs), which are applicable in Canada to the registered owners of the affected products. Equipment/appliance ADs are often only distributed to our regional offices because the owners of aircraft affected by this type of AD are not generally known.

TC has received the following new ADs on equipment in the last three months. AMEs and operators of the affected products are encouraged to obtain further information or a copy of the ADs from their regional TC office, their local TCC, their PMI, or from the Civil Aviation AD website at: http://www.tc.gc.ca/aviation/applications/cawis-swimn

PARACHUTES DE FRANCE	F-2004-015R1	FR	F-2004-015 is cancelled by Revision 1. Superseded by DGAC AD F-2007-025.
PARACHUTES DE FRANCE	F-2007-025	FR	ATA 25 - Equipment/Furnishings -Temporary grounding of reserve canopies.
PARACHUTES DE FRANCE	F-2007-026	FR	ATA 25 - Equipment/Furnishings - Reserve parachute – Mandatory packing modifications slider and steering lines replacement.
MICROTURBO	2007-0294	EU	Airborne Auxiliary Power – APU Turbine Wheel Life Limit – Reduction
SWISS AIR-AMBULANCE	HB-2007-295	SW	Horizontal Net SRFW HN1 - Removal from Service
STC SR00981LA (AEROM)	2007-12-23	US	Inspection for cracks of the landing gear strut assembly.

SUSPECTED UNAPPROVED PARTS

During the previous quarters there were no Service Difficulty Reports (SDRs) received that indicated any suspected unapproved parts.

In Canada, SUPs should be reported in accordance with Canadian Aviation Regulation (CAR) 591.01, indicating your suspicion of an unapproved part using WSDRS on the Internet at <u>www.tc.gc.ca/wsdrs</u> or on a regular SDR (24-0038)r2 form.

FAA SPECIAL AIRWORTHINESS BULLETINS (SAIBS)

A Special Airworthiness Information Bulletin (SAIB) is an information tool that alerts, educates, and makes recommendations to the general aviation community. It is non-regulatory information and guidance that does not meet the criteria for an Airworthiness Directive (AD). http://www.faa.gov/aircraft/safety/alerts/SAIB/

SAIB#	Manufacturer	Subject	Issue Date dd/mm/yy
CE-08-06	Schleicher, Alexander, GmgH & Co.	Engine Fuel and Control	2007-11-14
SW-08-05	MD Helicopter Inc.	MD900 Main Rotor Blade Retention Bolt Check	2007-11-14
SW-08-04	Eurocopter Deutschland GmbH	Night Vision Imaging System installation on BK -117C-2 Helicopter	2007-11-09
SW-08-03	Turboshaft-powered	Recommendations for Rotorcraft During Icing Conditions	2007-11-08
CE-08-02	Cessna Aircraft Company	Flight Controls	2007-11-02
CE-08-01	Burkhart Groβ Luft-Und	Fuselage, Wing, & Stabilizer Structure	2007-10-18
CE-07-44R1	Amateur-Built	Doors	2007-10-16
CE-01-41R2	Cessna Aircraft Company	Flight Controls	2007-10-01
NE-07-53	Propeller assemblies	Propeller assemblies overhauled by Desert Aircraft Blade Service	2007-09-25
NE-07-54	Microturbo	Microturbo TRS 18 Series Turbojet Engines	2007-09-24
NE-07-52	Smoke Detectors	Lavatory ionization fire/smoke detectors	2007-09-24
NM-07-55	Gulfstream Aerospace Corporation	Air Conditioning – Pack Inlet Valve Operation	2007-09-24
CE-07-50	Diamond Aircraft Industries GmbH	Doors	2007-09-20
CE-07-51	Diamond Aircraft Industries GmbH	Engine	2007-09-20
NE-07-49	Lycoming Engines	Fuel Injector Tube Assembles and Support Clamps	2007-09-20
NE-07-42R1	Turbochargers or related control components	Components Overhauled by Statesville AeroTech Services	2007-09-18

FAA UNAPPROVED PARTS NOTIFICATIONS (UPNs)

Unapproved Parts Notifications (UPNs) are published by: FAA, AIR-140, P.O. Box 26460, Oklahoma City, OK 73125. They are posted on the Internet at: http://www1.faa.gov/avr/sups/

No. 2007-00139 issued November 7, 2007

AFFECTED PRODUCTS

Wheel and brake assemblies and aircraft accessories.

PURPOSE

The purpose of this notification is to advise all aircraft owners, operators, manufacturers, maintenance organizations, and parts suppliers and distributors regarding improper maintenance performed on wheel and brake assemblies and aircraft accessories used on large aircraft.

BACKGROUND

Information received during a Federal Aviation Administration (FAA) suspected unapproved parts investigation revealed that between March 2006 and March 2007 Aerospace Precision, Inc. (P17R), located at 2851 Evans Street, Hollywood, FL 33020, improperly maintained and approved for return to service various aircraft accessories and wheel and brake assemblies contrary to the regulations. Aerospace Precision, Inc., holds FAA Air Agency Certificate No. P17R189O, with limited accessory rating.

Evidence indicates Aerospace Precision, Inc., approved wheel and brake assemblies and accessories for return to service that were not maintained in compliance with the manufacturer's maintenance manuals or other data acceptable to the FAA. Discrepancies noted in Aerospace Precision Inc.'s practices included, but are not limited to, the following:

- Approving for return to service wheel and brake assemblies and accessories that had not been inspected and tested in accordance with current nondestructive testing (NDT) manual.
- Using an NDT technician who was previously trained at another repair station, but not properly certificated to perform NDT inspections at Aerospace Precision, Inc.
- Failing to maintain recordkeeping requirements.
- Failure to use equipment, tools, and material that are recommended by the manufacturer and acceptable to the FAA.

Approving for return to service articles that had not been maintained in accordance with the current manufacturer's maintenance manuals or methods otherwise acceptable to the administrator.

RECOMMENDATIONS

Regulations require that type-certificated products conform to their type design. Aircraft owners, operators, manufacturers, maintenance organizations, and parts supplier and distributors should inspect their aircraft, aircraft records, and/or parts inventories for any wheel and brake assemblies and accessories that were approved for return to service by Aerospace Precision, Inc., between March 2006 and March 2007. If any wheel and brake assemblies or aircraft accessories are found installed on aircraft, appropriate action should be taken. If any are found in existing inventory, it is recommended they be segregated to prevent installation until a determination can be made regarding each part's eligibility for installation.

A *partial* list of parts that were approved for return to service by Aerospace Precision, Inc. can be found at:

http://www.faa.gov/aircraft/safety/programs/sups/upn/ media/2007/UPN 2007_0139.pdf

FURTHER INFORMATION

Further information and guidance regarding the abovereferenced wheel and brake assemblies and aircraft accessories can be obtained from the FAA Flight Standards District Office given below. In addition to all the above recommendations, the FAA would appreciate any information concerning the discovery of the wheel and brake assemblies or aircraft accessories from any source, the means used to identify the source, and action taken to remove the wheel and brake assemblies or aircraft accessories from service.

This notice originated from the FAA South Florida FSDO, 1050 Lee Wagener Blvd., Suite 201, Fort Lauderdale, FL 33315, telephone (954) 635-1300, fax (954) 635-1260. *****

SERVICE DIFFICULTY REPORTS	JLTY	REPORTS				
LEGEND						
JASC Joint Aircraf	ft Syste	Joint Aircraft System Code number defining assembly/system/component	system/component			
SDR NO. TCA assign	ed SDI	TCA assigned SDR control number - please quote in ar	in any correspondence or inquiries			
RGN TCA region of PAC = Pacific	ı of SD ific	TCA region of SDR submitter: PAC = Pacific PNR = Prairie and Northern	ONT = Ontario VAR = More than one Region	Quebec ATL = Atlantic	NCR = Ottawa (HQ)	a
Make/Model	JASC	Part Name	Part No.	PART CONDITION	Sdr No.	RGN
AIRCRAFT					l	
AERO COMMANDER 690 695A	3220 3220	Shaft Drag Brace Bolt	750069501 7500761	Cracked Cracked	20071010011 20071022005	PAC NCR
AEROSPATIALE AS 350B2 ATR 42 300 ATR 42 300 ATR 42 300	7921 6410 3246 3200	Blower Rotor Tail Rotor Tie Bolt ECU	350A5310520351 355A12004008 MS2125006038 7898435010	Unserviceable Unbalanced Broken Failed	20071120007 20071115005 20071024009 20071109005	PAC PAC PNR PNR
AIKBUS A319 114 P. T. C. ANTITTONI	2330	Harness wire (floor)	1	Broken	20071031009	ONT
BAE - (KAY 1HE UN) BAE 125 800A HS 125 700A HS 125 700A	2810 2820 3211	Vent Float Valve Fuel Tube (Assembly Metal) Nose Landing Attachment	257PV6589A 30733341 25FN1609	Defective Chafed Cracked/Vibrating	20071016021 20071106011 20071011007	QUE QUE ONT
BAE - UK 3112 3112 BAE 146 200A <i>BEEFCU</i>	2910 3210 3210	Main Hydraulic Return Line Bracket Directional Link	137311D1233 137414B100 200915254	Unserviceable Broken Unserviceable	20071026008 20071012001 20071108007	PNR PNR ATL
B22001 1900C 1900D 1900D 1900D 1900D A100 B100 B200 B200 B200 B200 B200 B200 B	3211 3260 2711 7500 7500 3210 3233 3233 3233 3233 3233	Brace Switch Trim Tab Actuator LH Flap Drive Cable Bleed Air Hose Torque Knee Flap Track Landing Gear Motor Bolts Inboard Aft Roller Area Pressure Gauge Actuator	998100287 1CH25 1295210325 1013800005 12991003315 50810323 501600183 1153800025 10310300 9536830 11013880141	Broken/Cracked Failed Seized Sheared Cracked Cracked Cracked Cracked Deployed New Worn Leaking Failed	20071114009 20071102006 20071010008 20071010008 20071010002 20071030002 2007101001 20071029005 2007101001 2007101000 20071011010	ONT PNR PNR PNR ONT ONT ONT ONT PNR PNR PNR NCR

Feedback 01/2008

RGN	PAC	PNR PNR PNR PAC ONT ONT ONT	NCR ATL PAC	PAC PAC PAC PAC PAC PAC	QUE ATL ATL ATL ATL ATL NCR NCR NCR NCR NCR NCR NCR NCR NCR NCR	OUE PNR OUE OUE OUE
Sdr No.	20071015005 20071031008	20071024002 20071023008 200711024005 20071120004 20071120004 20071001010 2 SDRs 20071030001	20071010005 20071026003 20071102013	20071026011 20071026009 4 SDRs 20071029009 20071106003 20071106003	20071017001 20071012003 2 SDRs 2 SDRs 2 SDRs 2007105001 2007105005 20071105005 20071103001 2 SDRs 20071103001 2 SDRs 2 SDR5 2 SD	20071003004 20071001007 20071017002 20071017002 200710129007 20071115006 20071011006 20071012001
PART CONDITION	Damaged On condition	Cracked Worn Worn Defective N/A to assembly Cracked Failed FOD	Cracked Corroded Unserviceable	Corroded Corroded Chafed Contaminated Scratch Chafed	Failed Failed Unserviceable Frozen Chafed Cracked FOD Worn/Open Unserviceable Unserviceable Unserviceable Cracked Broken Rod End Broken	Missing Cracked Cracked Leaking Sparking Sparking Unserviceable Cracked
PART No.	CH5339906 950476	CL422501 206010743013 41103750017 206040272101 476411731 206310105101 206040078103 204076437003	204011250001 204001363025 476206283	453A26113 147A55067 112A410209/10 273A45141 146AS323110 112A4102209	A3579101 442324 855D10011 852D10019/21 CA447 600330061 601R6000267 BJ10021003 601R120474 22850080 21188002 CF343B1873574 601R3303312 NP13932111	21587525 33130131 215003268 21564002 DAT4864A 6047T74P13 22852402115
PART NAME	Lead, Ignition Bearing Cone	Clutch Assembly Pitch Change Mechanism Servo Drive Shaft Boot Thrust Plug/Shim T/R Feathering Bearing Nut Check valve	Main Rotor Blade Cyclic Tube Bearing	Cap Strip Floor Beam Wing Skin Shaft Assembly Skin Wing Skin	Generator Hydro-Mechanical Unit (HMU) BPSU Flap Actuator Fuel Shroud Radome Assembly Engine Assembly Wing Rib T/R Cowling ECU Engine Assembly RH Window Window Window Window Window Window	Adjusting Plate Piston Spar Cap Fuel Cell Rudder Bearing Bus Bar Fuel Control Unit Tube Assembly
JASC	7420 3246	6310 6310 6730 6215 6400 6420 6420 6420 2915 2915	6210 6700 6320	5315 5315 5730 5730 5730 5330 5330	$\begin{array}{c} 2434\\ 7321\\ 2750\\ 2755\\ 2752\\ 3050\\ 3244\\ 7110\\ 3200\\ 5754\\ 7110\\ 3200\\ 5610\\ 5610\\ 2400\\ 2730\end{array}$	3213 5246 5711 2810 5544 7321 7321
Make/Model	B300 B99	BELLI LEALKON - CAIVADA 206B 66 206B 66 206B 66 206B 66 206L 1 66 412EP 66	BELL IEXIRON - USA 204B 212 47G2 POENC	БОЕЛИС 737 790 737 790 737 790 737 990 737 990 737 990	ED 100 1A10 ED 100 1A10 ED 100 1A10 CL600 2B19 (RJ100) CL600 2D24 (RJ900) CL600 2D24 (RJ900) CL600 2D24 (RJ900) CL600 2D24 (RJ900)	CL215 1A10 CL215 1A10 CL215 1A10 CL215 1A10 CL215 6B11 (CL415) CL215 6B11 (CL415) CL215 6B11 (CL415) CL215 6B11 (CL415) CL200 2B16(601 3A) CL600 2B16(601 3A)

RGN	ONT PAC PNR PNR PNR PNR PNR PNR PNR PNR PNR PNR	NCR	QUE	PAC PAC PAC PAC PAC PNR PNR PNR PNR PNR PNR PNR PNR PNR PNR	ATL
Sdr No.	20071002004 20071002003 20071019002 20071102012 20071109007 20071104006 2 SDRs 20071114003 200711114003 200711114000 200711114010 200711011002	20071009003	20071026004	20071024006 20071026010 200710108003 20071108003 20071102007 20071107005 20071115004 20071115004 20071115004 20071115004 20071012000 20071105003 200710020001 20071012000 20071102001 20071102001 20071102001 20071102001 20071102001 20071102001 20071102001	20071115008 20071102004
PART CONDITION	Chafed Overheated Screws broken in Elongated hole Separation Broken teeth Overheated Broken Worn Loose Corroded Failed Cracked Cracked New	Cracked	Unserviceable	Leaking Corroded Cracked Cracked Cracked Cracked Corroded Life Used Used Used Used Used Used Cracked Cracked Cracked Cracked Cracked Cracked Cracked Cracked Cracked Cracked Severed Cracked Cracked Severed Cracked Severed Cracked Cracked Severed Cracked Severed Cracked Severed Cracked Severed Cracked Severed Cracked Severed Cracked Severed S	Seized Damaged
Part No.	NAS680A08 510128 5412024 Unknown CM358930 991027114 55657544 MS21090348 11128 300SGL129Q2 523809 78023061	71500	10150	PV332010 C2US1285A Unknown C2TP57 C4WM1190 18271 C6FSM181421 C6FSM181421 C6FSM18229 C6W11509 L814749 L814749 L814749 L814749 L814749 L814749 L814749 L814749 L814749 L814749 S2760059101 82920010265 82970010149 DSC1896 82740162001 NAS1304 20011610 473901 699018002	HF3M 30200530001110
PART NAME	Scat Hose Wire Landing Light Switch Nut plate Bracket Main Gear Leg Bushing Gear Switch/Circuit Breaker Mixture Control Lever T/R Stow/Deploy Switch Rim Starter/Generator Rib Muffler Magnetic Seal	Seat Frame	Steering Assembly	Fuel Shut-Off Valve Link Assembly Lower Attach Fitting Stab Spar Bracket DC Power Supply Control cables Adaptor Assembly Aft Adaptor Plate Bolt Rib Bearing Wiring Wiring Wiring Adaptor, Actuator Mount Bearing Wiring Hydraulic Tube Hydraulic Pressure Line Check Valve Link Rod Bolt Hydraulic Pressure Line Check Valve Link Rod Bolt Filter Unions Nose Lock Harness RH Propeller Electronic Unit (PEC) Cargo/Baggage Doors LH Engine	Rod End Bearing Tube
JASC	8530 3397 5730 27730 27730 3211 27730 7602 7602 7602 7602 7830 7830 7830 7830 7830 7830 7830 7830	5347	3250	D 2810 5511 5511 5511 5511 5511 5511 5511 5	7602 3245
Make/Model	CESSNA 152 152 172K 172P 172P 172S 208B 208B 208B 208B 208B 208B 208B 208B	CHAMPION 7ECA	CONVAIR - CANADA 440 DEMANTITAND CANAD	DEHAMILLAND - CANADA DHC 2 MKI DHC 6 300 DHC 8 102 DHC 8 400	DA 20 C1 DA 20 C1

AME	Part No.	PART CONDITION	Sdr No.	RGN
sembly ake	MS21919WDG9 7458271 MS24693C273	Loose Broken Wrong Size Fuel Leaks Seized	20071029002 20071025010 20071107004 20071101003 20071101003	SUE SUE
ring	6894171 V40G3911AA	Making metal Cracked	20071022006 20071025001	ONT
Plug itrol	BM1021 7050A4673006	Shorted No Lock/Test	20071114007 20071005002	UNT ONT
le ing Arm Assembly tainer Accumulator cking	31032359 2752530001 31025731 223002/03 AN17722 AN363382	Failed Cracked Cracked Cracked Cracked Missing Wrong part	2 SDRs 2 SDRs 20071106009 20071106008 2 SDRs 2 SDRs 2 0071121002	ONT ONT ONT PNR ONT ONT
Frame	20A6050	Unserviceable	20071029008	PAC
Actuator shield	263000004 NP17820110	Faulty Cracked	20071108006 20071106001	UNT ONT
tion Loop		Failed	20071112003	PNR
r Pedal Bracket	369A75058	Broken	20071112002	PAC
Rudder Trim smbly	793500501 50028053	Substandard Failed	20071026005 20071026007	PNR NCR
c ~	66082484 231127416 BNC3 and BNC4 226K084	Burnt Cracked Reversed Delaminated	20071010004 20071004003 20071030005 20071106002	PNR NCR NCR PAC
khaust Silencer	L24266710000	Cracked	20071109003	ONT
e Indicator SW	43069	Serviceable	20071031003	ONT
ing b mbly e Connector : Control	5321012193 5PC41301001 9433772104 5714231557	Swollen Unserviceable Cracked New Pin Loose Lamp not illuminated	20071005003 20071114005 20071113007 20071105002 20071119002 20071119001	ONT ONT ONT ONT ONT ONT ONT

SC PART NAME		1 Clamp Assembly			0 Hose Core 1 Brake			0 Ejector Spring		0 Servo, Control			0 Fuer Ivozzie 0 Nose Steering Arm Assembly			1 Nut self-locking	1	1 Oil Cooler Frame	3 LH MLG Actuator 0 RH Windshield		0 Fire Detection Loop		0 LH Rudder Pedal Bracket	2 Actuator, Rudder Trim			1 Motor				0 Primary Exhaust Silencer		7 L Eng Fire Indicator SW		 Guiding Ring AOA Vane 			
JASC	•	2701	3242	1000	2820 3241	1	7414	0770	6370	6730		2410	6110	6110	2911	3241		7311	3233 5610		2600		2720	2722	3242		2421	3418	3234		7800		2697		3222 3418	5317	1410	1410 3412
MAKE/MODEL	EMBRAER	ERI 170 200 SU	ERJ 170 200 SU	ERJ 190 100 IGW	ERJ 190 100 IGW ERJ 190 100 IGW	EUROCOPTER DEUTSCH	BO105 S CDN BS 4	ELIROCODTER FRANCE	FC 130 R	EC 120 B EC 120 B	FAIRCHILD	SA227AC	SA227AC	SA227CC	SA227DC	SA227DC	GROß-WERKE	G 120A	<i>GULFSTREAM – ISRAEL</i> GULFSTREAM 200 GULFSTREAM 200	HAWKER SIDDELEY-UK	HS 748 2A	HUGHES	369D	ISRAELI INDUSTRIES 1124	1124	LEARJET	31	35A	45	MORAVAN	Z242L	PLAGGIO	P180 AVANTI	PILATUS - SW	PC 12 45 PC 12 45	PC 12 45	PC 12 45	PC 12 45 PC 12 45

Make/Model	JASC	Part Name	Part No.	Part Condition	Sdr No.	RGN
PC 12 45 PC 12 45	$\begin{array}{c} 3110\\ 3160 \end{array}$	Electronic Attitude Director Indicator Display (Electronic Horizontal Situation Indicator	0660312525 066031252500		20071119003 20071120005	UNT ONT
PA24 250 PA31 PA31	3230 3700 2435		2124800 441CC7	Jammed Noisy Failed	20071010007 20071023007 20071114001	NCR NCR PAC
PA31 PA31 350 PA31 350	7720 5347	e Belt Assembly	10816811	Unserviceable Broken	20071030008 20071015001	NCR ATL
PA31 350 PA31 350 PA31 350 PA31T	2435 8530 3210 3000	Case Fwd Crankcase Oil Seal MLG Side Brace Inlet Ice Screen	LW15628 4028400 50363007	Falled New Broken Cracked	20071004008 20071004008 20071001009 20071011008	PNR PNR PNR
ROBINSON						
R44 II R44 II R44 II	2913 2916 6310	Pump Assembly Hydraulic Reservoir Clutch Assembly	D5001 D2112 C0183	Replaced Venting Removed	20071022009 20071022010 20071022007	NCR PNR PNR
R44 II R44 II	6520 7314		B5631 C8187B	Leaking Failed	20071019001 2 SDRs	PNR NCR
R44 II R44 II	7800 6730	Exhaust/Heater System Servo	C16932	Failed Leaking	20071114004 20071004005	PNR PNR
ROCKWELL)		
	7720	ITT Indicator	850568503	Unserviceable	20071109006	PAC
SIKORSKY						
	2820 6320 2562	Hose Bearing Denlovable Beacon	7,63070E+12 SB2151107 50316	Collapsed Spalled Failed	20071016020 20071115007 20071024007	PAC PAC PAC
SWEARINGEN						
SA226TC	5610	Heated Windsheild	2719442004	Burnt	20071116001	PNR
ENGINES						
ALLISON						

										12 QUE		PNR			
	20071001006	2007101000	200710230(200710300(2007101500	2007103001	2 SDRs	20071015006	200711150(2007100101		3 SDRs	200710220(4 SDRs	
	Cracked	Cracked	Metal Contaminated	Damaged	Failed	Repairable	Failed	Out Of Adjustment	Cracked	Shut down		Leaking	Failed	Cracked	
	23037413	6870992	6894171	6898607		6899246		23065104	6871259						
	PT Outer Shaft	Combustion Case	Gearbox	#1 Compressor Bearing	Drive Spline	Oil Filter Housing	Engine Assembly	Fuel Control Unit	Compressor Tie Bolt	Engine		Fuel Pump	Bendix Servo Fuel Injector	Engine Starter	
	7250	7240	6320	7230	7323	7920	7931	7321	7230	7200		7314	7322	8011	
ALLISON	250-C20	250-C20	250-C20B	250-C20B	250-C20B	250-C20B	250-C20B	250-C20B	250-C20J	AE-3007A1	AVCO LYCOMING	IO-540-AE1A5	IO-540-AE1A5	IO-540-AE1A5	

RGN	ONT OUE PNR PNR PNR	ONT PNR	QUE QUE QUE VAR	PURE PURE PURE PURE PURE PURE PURE PURE	CLE CLE	CUE CUE CUE CUE CUE CUE CUE CUE CUE CUE	OUE OUE OUE
Sdr No.	20071003007 20071102005 20071017003 20071109011 20071022002	20071029010 20071120002	20071003018 20071003028 20071003022 20071016017 20071113005 4 SDRs	20071016018 20071016018 200710010111 20071016004 20071107002 2 SDRs 2 SDRs	20071016013 20071003016 20071003024 20071003027 2 SDRs 20071019004 20071019004 20071016008 2 SDRs 2 SDRs 20071016001 2 SDRs	20071016014 20071025009 20071025005 20071025005 20071024011 2 SDRs 20071003009 20071003019 20071003019 20071016009 20071016009	20071003011 20071003025 20071025008 20071016002 20071016007
Part Condition	New Cracked Cracked Failed Excess Clearance	Used Compressor stall	Smoke Blocked Blades damaged Damaged Unserviceable Unserviceable	Fracture Seized Damaged Failed Blades damaged Seized Fractured	Damaged Fracture Damaged Undetermined FOD FOD Fractured Leaking Smoke Fractured	Faulty Fractured Defective O-ring Low FOD Leaking Defective FOD Unserviceable Haulty	Undetermined Replaced Undetermined Internal damage Damaged
Part No.	LW13383 LW167035C 104001667 LW13870 LW10292	63E951	32448809	310926302	PT08001A01	3034677	311608001
PART NAME	Baffle Cylinder Assembly Impulse Cam Assembly Cylinder Assembly Crankshaft Idler Gear	Housing LH Engine	Engine Scavenge Oil Strainer Engine Combustion Chamber Liner Hydro-Mechanical Fuel Control Unit Hydro-Mechanical Fuel Control Unit	Compressor Turbine Blade Engine Assembly Large Exit Duct Engine Engine Fuel Control Drive	Engine Turbine Blade Power Turbine Blade Engine Engine Towershaft Valve, Tube, Oil Filler Generator Pad Adapter Engine Turbine Blade	Engine Oil Pressure Turbine Blade Engine Oil Pressure Engine Oil Pressure Engine Tube Oil Pressure Electronic Engine Control Engine T6 Thermo Coupler Fuel Control Unit	Engine Engine PEC Engine Bearing No.3
Jasc	7920 8530 7414 8530 8530 8520	2821 7230	7200 7931 7250 7240 7310 7310	7250 7261 7200 7200 7200 7200 7200	7250 7250 7250 7200 7200 7200 7200 7200	7931 7250 7931 7931 7931 7931 7200 77200 7720 7320	7712 6120 7200 7250
Make/Model	IO-540-L1C5 O-235-L2C O-320-H2AD O-540-F1B5 T1O-540-J2BD	CEAPERAL ELECTRIC CF34-3A1 CT7-9B DP 4TT F& WHITWEV CANADA	JT15D-4 JT15D-4 JT15D-5 JT15D-5A JT15D-5A JT15D-5D PT6A-114A PT6A-20	PT 6A-27 PT 6A-27 PT 6A-28 PT 6A-34 PT 6A-41 PT 6A-41 PT 6A-42 PT 6A-42 PT 6A-42	PT 16A-65AG PT 6A-65B PT 6A-67D PT 6A-67D PT 6A-3DF PT 6T-3DF PW 118 PW 120A PW 120A PW 120A PW 121 PW 121	PW121 PW123 PW123 PW123D PW124B PW124B PW125B PW125B PW125B PW127E PW127E	PW127F PW150A PW150A PW150A PW206A

Part Name
Engine Engine Fan Blade Electronic Engine Control Fuel Control Unit Engine
T4 Disk
Cylinder Cylinder Crankrase Assembly
(
HP 1 T/Blade Impeller LP
D1 I inc
2
Engine Intake Tube
intake 1ube Cylinder Assemblyy Exhaust Rocker Shaft
Hydro-Mechanical Unit
Propeller Blade
Propeller Bushing
1 orth Somiran
Sund
E.L.T.
Pitch Horn
Support
Alternator

Short-circuited 2071109012 NCR Failed 20071001008 PNR Replaced 20071010019 PNR SB2151107 Spalled 20071012004 PNC SB2151007 Spalled 20071012004 PNC SB2151007 Spalled 20071012004 PNC SB2151007 Spalled 20071010010 PNC MA18 Wrong Part 20071010010 PAC MA18 Wrong Part 20071010010 PAC Cracked 20071010000 PAC PAC	
Failed 20071001008 Replaced 20071016019 Spalled 20071012004 Undamaged 20071012004 Incorrectly installed 4 SDRs Wrong Part 20071019005 Cracked 20071101004	
Replaced20071016019103Spalled20071012004Undamaged20071010010Incorrectly installed4 SDRsWrong Part20071019005Cracked20071101004	
103Spalled Undamaged20071012004 20071010010103Incorrectly installed4 SDRsWrong Part20071019005Cracked20071101004	
103 Optimized Undamaged 20071010010 Incorrectly installed 4 SDRs Wrong Part 20071019005 Cracked 20071101004	
Incorrectly installed4 SDRsWrong Part20071019005Cracked20071101004	
Incorrectly installed 4 SURS Wrong Part 20071019005 Cracked 20071101004	
Wrong Part 20071019005 Cracked 20071101004	
20071101004	

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Canadian Aviation Regulations (CARs) www.tc.gc.ca/civilaviation/regserv/Affairs/cars/menu.htm

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Web Service Difficulty Reporting System (WSDRS) www.tc.gc.ca/aviation/applications/cawis-swimn/logon-wsdrs-cs16101. asp?lang=E&rand=