Issue 3/2003



feedback

Canadian Aviation Service Difficulty Reports

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hangar *noise*

A Message for Aircraft Maintenance Personnel

SDR Logic Chart - Revision 1....a new look

The SDR Logic Chart had been in circulation for approximately two years and has been very positively received by operators and maintainers. Revision 1 incorporates minor changes with the objective of improving and clarifying some of the modified text.

Submitter Type Section:

- → New sub-headings of Mandatory and Voluntary Reporting to better emphasize applicable CAR requirements;
- → Flight Training Units and Private Operators have been added to include their regulatory obligation. General Aviation operators have also been added to invite voluntary SDR submissions. For convenience, the applicable CAR references are included.

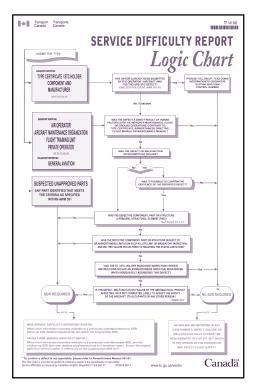
Center Column:

→ Several center boxes have minor changes including a new box that specifically addresses Principal Structural Elements (PSE), including the reference to Advisory Circular AC25-571-1C:

Principal Structural Element (PSE) is an element that contributes significantly to the carrying of flight, ground, or pressurization loads and whose integrity is essential in maintaining the overall structural integrity of the airplane.

Bottom Right Column:

→ An asterisk * has been added to the "NO SDR" box. This "*" directs the submitter to AWM 591.01 for confirmation that no SDR is required.



This SDR Logic Chart is provided as convenient guidance material to assist the user in identifying reportable service difficulties.

Note:

Revision 1 SDR Logic Chart (TP14134B) has a two-sided bilingual format and a new background appearance. SDR Logic charts are available at your local Transport Canada Center (TCC) or can be requested via e-mail address: SDRS@tc.gc.ca.

For more information or copies of **feed** *back* or other Civil Aviation publications, call 1-800-305-2059 or visit our web site at www.tc.gc.ca/civilaviation/certification. To ensure continued delivery, send any address changes to: Transport Canada, Civil Aviation Communications Centre AARA, Place de Ville, Ottawa, ON, K1A ON8.

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Cette publication est aussi disponible en français.

fixed wing

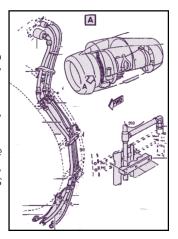
AIRBUS A320

Hydraulic Hose Chafed

A hydraulic hose was found chafed through on Engine 1 green hydraulic system pump pressure transducer of an A320. This resulted in substantial fluid loss, emergency decent and a subsequent unscheduled landing.

Most power plants have a myriad of lines for the various engine and airframe systems, located out of necessity in close proximity to various other lines and components.

The SDR program continues to receive many reports of line failure due to inadequate clearance to adjacent structure. By ensuring adequate clearance and/or protection, both at initial installation and during maintenance, the risk of this costly failure is negated.



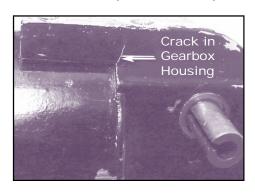
SDR # 20030506006

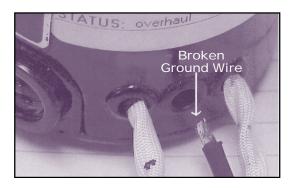
Flap Motor Defective

BEECH 58P

Subsequent to the installation of a newly overhauled flap motor (P/N 3536415157); the AME noticed intermittent flap indications during functional checks.

Upon removal and investigation of the suspect flap motor; the source of the problem was traced to a faulty/loose internal wire. A further inspection of the flap motor revealed an obvious crack in the gearbox housing.





Transport Canada reminds maintainers to carefully examine all aeronautical parts for defects and also for traceability before installation. The SDR data base lists many other similar reports of this nature.

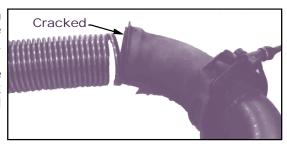
BEECH C90A

Anti-Ice Boss Cracked

SDR # 20030115006



During a routine inspection on a Beech C90A, the anti-ice boss on the LH exhaust stack (P/N 109-950000-1) was found broken off with the anti-ice flex tube. The flex tube was also cracked around the lower elbow.

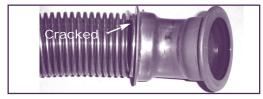


A heads up for maintainers when inspecting this area.

BEECH C90A Anti-Ice Tubes Cracked

SDR # 20021125012

During inspection on a Beech C90A, the R/H engine cowl was removed and the anti-ice flexible tube, PN 90-910100-17 that routes exhaust air to heat the engine cowl intake was discovered cracked. The crack extended approximately 1/3 of the circumference of the tube.



Time Since New (TSN): 124.6 hrs

Manufacturer's Service Bulletin 71-3142 was issued to reduce cracking in this area. However, cracks have been reported subsequent to compliance with SB71-3142.

Transport Canada recommends increased vigilance in this area for cracks.

CESSNA 150 SDR # 20030423006

Horizontal Stabilizer Cracking

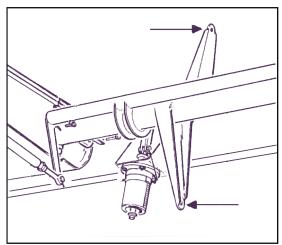
Leading Edge Rib assemblies on a Cessna 150, part numbers 0432001-46 & 0432001-6, are very difficult to see. The use of a mirror through the horizontal spar and rib lightening holes is the only way to visually inspect for cracks along the tabs.

The submitter believes the ribs are cracking due to improper ground handling. The practice of pushing the aircraft's tail down by its spar in order to rotate the aircraft in position may be damaging these ribs. Both ribs were replaced with new.

Small aircraft may be moved by hand, however damage can be inflicted on these aircraft by improper procedures. Be sure to use a tow bar.

DE HAVILLAND DHC 2 MKI Cable Attach Point Elongated

SDR # 20030318009



During inspection, the elevator torque tube, part number C2T7A, lower cable attach point was found to be elongated.

TSN: 21,878 hours.

Frequently, this sort of defect is discovered only when other maintenance requires the removal of an associated component. Diligent inspection was necessary to turn up this defect that may otherwise have gone unnoticed.

Defects occurring within a cable loop may be indicated by an unexplained change in cable tension or the inability to obtain constant cable tension readings when rigging.

To accurately inspect for this kind of defect, control cables must be loosened in order to detect any play in the attachment ends.

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DE HAVILLAND DHC 7 102

SDR # 20030514008

Wiring Chafed

During heavy maintenance inspection on DHC 7 102, two wires were found (2610-300 13a22 and 2610-30006a22) chafed and broken at deteriorated clamps in the lower aft firewall area. The rubber cushion had torn away allowing the metal clamp to wear the wires. These wires are for the fire loop system on the #3 engine, which was working when the aircraft was taken down for maintenance. Only the support clamps were holding the wires together.

The fire loop system has been snagged many times in the past; it would fail the pre-flight test three or four times per year, mostly on cold winter mornings. It was usually repaired by cleaning/replacing the connections in the fire loop. It is suspected that the bad connection in these chafed and broken wires may have increased the resistance in the system and caused the test to fail in the past.

This defect illustrates wiring damaged as a result of a deteriorated clamp cushion. It was found during inspection of an aircraft approaching the "aged stage" of its life (20 years).

ISRAEL ASTRA SPX

Duct Cracked

SDR # 20030424004

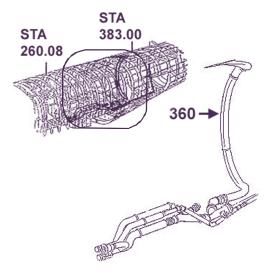
During a 1000-hour inspection, the aft baggage compartment interior was removed, allowing access to the ram air duct. This plastic duct is located on the right side tucked tightly against the fuselage skin at STA 350.

During inspection, a 10-inch circular crack was found in the duct (P/N 25W783016011) adjacent to the fuselage skin at approximately Stringer #6R (counting down from upper stringer).

The nature of this crack allowed the duct to open up when internal pressure was applied. This crack was only evident by feel, or using a flashlight and mirror.

The duct is used to clear the fuselage interior in the event of in-flight fire/smoke. Replacement of the duct was required.

Israel Aircraft Industries has introduced a new part number duct.

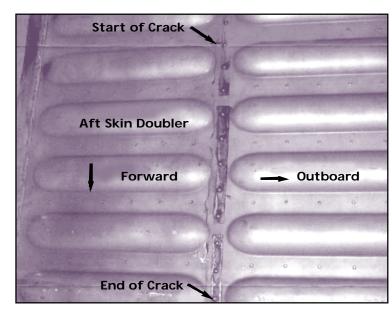


PIPER PA23 250 SDR # 20030409015

Down Lock Mechanism Failed

When the pilot selected landing gear up, the gear did not retract. The submitter discovered that the down lock mechanism PN 1764500 was contaminated with dirt. The landing gear was cleaned and lubricated, and gear retractions were successfully carried out. The aircraft was returned to service.

There are many Service Difficulty Reports in the Transport Canada database identifying deficient landing gear cleaning and lubrication. AC 43.13 emphasizes that cleanliness is important during and after lubrication. Cleanliness may prevent unnecessary dirt accumulation and prevent component malfunctions.



PIPER PA 28 140 SDR # 20030214006 Wing Walk Doublers Cracked

The doublers on the bottom of the wing walk skins PN 00-62061-002/804 just outboard of the stamped stiffeners were severely cracked. The aft skin crack was 42cm in length. The forward skin had three cracks 5cm in length.

This area is not normally accessible. However, this aircraft (PA28 140) has an extra inspection hole installed from a previous Service Bulletin. The submitter advises that his area be inspected with a boroscope, especially on higher time aircraft.

SDR # 20030422002

TSN: 9402 hrs

PIPER PA 31T2 SDR # 20030313003

Pneumatic Regulator Valve Failed

On approach of a Piper PA31T2, the crew noticed multiple avionics failures. The VOR signal strength was very weak; #1 radio and most navigation equipment were inoperative. The pneumatic pressure was noted as being at 21 PSI, when it should have been at 18 PSI. The pneumatic regulator valve PN 584228(1H5210) had failed.

This failure caused hot air to be released at the pressure relief valve, which is located in close proximity to an antenna coaxial cable bundle. The hot air hardened and damaged the cables, causing the avionic failures mentioned above.

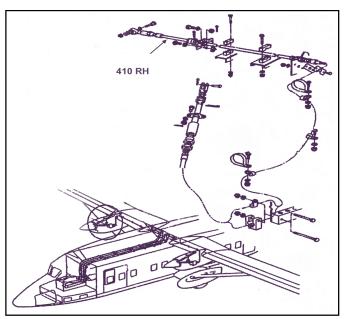
SHORTS HARLAND SD 360

Teleflex Cable Broken

On approach the pilot could not get R/H engine power low enough for approach power setting. The aircraft landed at a higher than normal power setting and after landing, neither beta nor reverse selections were possible. Maintenance personnel discovered that the power lever movement in the cockpit was limited in the forward direction, and no movement in either direction was visible at the engine.

Investigation revealed that the R/H power control Teleflex cable (P/N A55 2326 Post Mod A8457) had broken inside its conduit. Location of the break was at a 180-degree bend in the cable just below the engine control cam cluster. The problem may have been compounded by the lower cowl interference with the cable conduit when the cowl was in the closed position, causing additional stress in the area of the bend. It was noted that the cowl interfering with the cable would be difficult to see with the cowl closed.

There was no significant damage to the exterior conduit so this defect would unlikely be detected in advance.



This SDR stresses the importance of proper routing and ensuring adequate clearance is obtained with the adjacent structure.

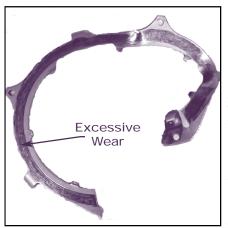
rotor*craft*

AEROSPATIALE AS 350BA

SDR # 20030429002

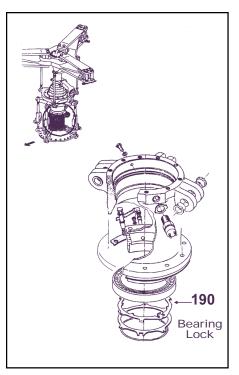
Main Rotor Mast, Bearing Lock Broken

Main rotor gearbox was removed for metal contamination and sent for overhaul. Upon disassembly, the bearing lock (P/N 350A37-1183-22), located between the lower bearing inner race and the epicyclic carrier

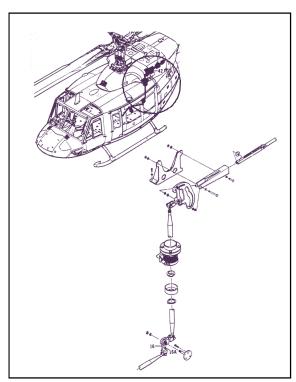


shoulder, was found broken into multiple fragments. This caused extensive damage to the epicyclic and contaminated the reduction gearbox assembly.

A service history review of past failures of the subject lock indicates that cracks may be a result of inadequate torque pre-load on the mast bearing lock. When replacing any one of the two mast bearings, the thickness of the shims must be adequate to obtain specified pre-load. If not, looseness and unwanted vibrations may result causing damage.



Transport Canada reminds maintainers to verify the bearing shim stack-up and to carefully check the subject lock for wear whenever mast seals require replacement.



BELL 212 Bellcrank Shear Rivet Sheared

RDS # 20030416003

While on approach, the droop compensator system of a Bell 212 failed to respond and engine rpm increased as the collective lever was lowered. The pilot adjusted engine rpm using the electrical trim system (beep switch).

Subsequent inspection found that the engine governor control bellcrank shear rivet (P/N MS20470B3) had sheared. No binding could be found in the governor control system linkages. Failure of the shear rivet is likely the result of fatigue.

EUROCOPTER EC 120 B SDR # 20030409012

Cabin Smoke

During cruise flight, smoke was noticed in the cabin. Shortly after an immediate and uneventful landing, all electrical power was lost (including battery power).



Ground wire to metal on top of vertical plate

over 1000 hours time since new (TSN). A similar occurrence had previously occurred on a foreign registered EC 120 B rotorcraft.



Light Ancillary Control Unit (LACU)

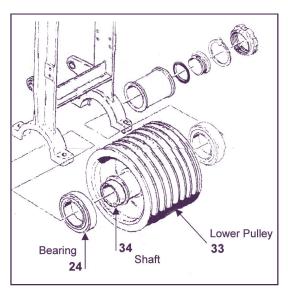
Follow-up investigation revealed that the wire harness leading directly to the anti-collision light on the upper vertical fin had chafed through causing a direct electrical short to the Light Ancilliary Control Unit (LACU). The wiring was repaired and the rotorcraft returned to service. This particular rotorcraft has been in service for

There are numerous inputs into the LACU such as landing/strobe/position lights, master battery "on", generator switch, fuel pump, pitot heat, rotor RPM, fire test, etc.

Transport Canada recommends that all EC 120 operators inspect this area for similar wire chafing pending further action from the manufacturer.

SCHWEIZER 300C SDR # 20030425008

Main Rotor Drive, Lower Pulley Shaft Bearing Slipped



Drive Assembly and Main Rotor Transmission

While carrying out a detailed inspection of the belt drive (300-hour inspection), the engineer noted that the inner race of bearing, P/N 269A5050-85, had slipped on the shaft, P/N 269A5497-9, as indicated by the slippage marks. Upon disassembly of the "H" frame, it was noted that the bearing could be hand rotated on the shaft. The assembly was removed from service because the shaft and bearing are press fitted.

Investigation revealed that some shafts might have been manufactured incorrectly by the OEM with a very slight taper, which later resulted in the bearing becoming loose enough to rotate on the shaft. The OEM has a repair for tapered shafts, which is to grind and then nickel-plate the shaft.

Close attention to the slippage marks on the shaft should adequately detect any unwanted movement of the bearing.

heads up

BEECH 95B55 SDR # 20030507007

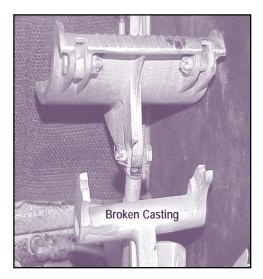
Rudder Pedal Cracked

During a flight control operational check during scheduled maintenance, the L/H rudder pedal PN 505243263 fractured at its attachment point. Oblong wear was discovered in the attachment area.

The wear was not easily noticed due to design and attachment to the brake master cylinder. Therefore removal of attaching hardware securing the pedal was required to inspect for any additional wear. A subsequent inspection of the remaining company Beech 95B55 aircraft revealed similar defects.

TSN: 6189.4 hrs

This rudder pedal part number is employed in various Raytheon/Beech aircrafts; pay close attention when inspecting this area.



equipment ADs

Transport Canada endeavours to send copies of new ADs which are applicable in Canada to the registered owners of the affected products. This type of AD is often only distributed to our regional offices because the owners of aircraft affected by equipment/appliance ADs are not generally known.

The following new equipment ADs have been received by Transport Canada in the last three months. Maintainers and Operators of the affected products are encouraged to obtain further information or a copy of the ADs from their regional TC office, their local TCC, their PMI, or from the Civil Aviation AD website at:

http://www.tc.gc.ca/civilaviation/certifcation/continuing/ad.htm

MANUFACTURER	AD NUMBER	ORIGIN	DESCRIPTION
ANJOU AÉRONAUTIQUE	2002-104(AB)R2	FR	SAFETY BELTS & RESTRAINT SYSTEMS TYPE 343-1 ANJOU AERONAUTIQUE SB 343-1-25-01
ANJOU AÉRONAUTIQUE	2002-105(AB)R2	FR	SAFETY BELTS & RESTRAINT SYSTEMS TYPE 343 SB 343-25-02
APIC	2002-453(B)R1	FR	AUXILARY POWER UNITEXHAUST TAILPIPE EJECTION CRACKS IN INNER LINER - APIC
BRUGGEMANN GMBH	2003-162	GY	EMERGENCY PARACHUTE- RETTUNGSFALLSCHIRM RFS II- OPENING MECHANISM - SBNR. LBA 001

AGING AIRCRAFT (PART II)



> Ramon (Ray) Raoux, P. Eng.

Background

Feedback Issue 1/2003 described the structural concerns and the subsequent maintenance activities that were developed for maintaining the airworthiness of aging aircraft. While the structural matters were being addressed in the 1990s, concerns were raised that the electrical systems in older aircraft might also need similar action. There had been a number of onboard fires, such as the Air Canada DC-9 near Cincinnati in 1983 and the fuel tank explosion in a B737 in Manila in 1990; however, these events did not lead to any high profile committees to study electrical systems. It was the 1996 explosion and crash of TWA B747 Flight 800 off Long Island, New York and the subsequent extended newsworthy investigation that put electrical systems and wiring under the microscope. In the USA, the White

House Commission on Aviation Safety and Security recommended to the FAA "In cooperation with airlines and manufacturers, the FAA's Aging Aircraft program should be expanded to cover non-structural systems". This recommendation spurred the

establishment in late 1998 of the Aging Transport Systems Rulemaking Advisory Committee (ATSRAC) to address, initially, electrical systems including wiring, connectors, and other components such as circuit breaker devices. The onboard cabin fire and crash of Swissair MD 11 Flight 111 near Peggy's Cove, Nova Scotia in September 1998 hastened the announcement of the creation of ATSRAC and added to the urgency for addressing electrical system concerns.

ATSRAC Membership

ATSRAC has representation from airworthiness authorities, manufacturers, airlines and other interested groups. The FAA, the European Joint Aviation Authority and Transport Canada represent the airworthiness authorities; Airbus, Boeing and Dassault represent aircraft manufacturers; AirTran Airways and Northwest Airlines represent the airlines. Some of the other ATSRAC members are NASA, the US Department of Defense, Air Line Pilots Association, International Association of

Machinists, Air Transport Association of America and General Aviation Manufacturers Association. In general terms, ATSRAC has been tasked to advise on regulations and supporting advisory material which may be necessary to assure that transport airplane non-structural systems are designed, manufactured, and maintained in a manner that will assure their safety throughout their service life.

The approach taken by ATSRAC to address aircraft wiring concerns was similar to the approach that was used to address aging aircraft structures. In the first phase, five working groups were formed and given specific tasks. One working group (WG) coordinated the inspection of wiring [in accordance with a specific inspection protocol] in over 100 large transport-category aircraft while they were undergoing heavy maintenance checks, in order to

determine the general condition of the wiring and associated hardware. Wire-related service bulletins (SBs) and airworthiness directives (ADs) were reviewed to determine if SBs should be mandated or ADs amended to incorporate terminating actions.

Intrusive inspections [where sections of wire bundles are removed] were performed on six aircraft recently retired from service and the removed wire bundles were sent to laboratories for further testing, primarily to categorize the effects of age-related service on the insulation characteristics of the wires. Another WG evaluated the effectiveness of maintenance programs with respect to wiring systems. A review of training programs and Electrical Standard Wire Practice Manuals (ESWPMs) were scrutinized by two other WGs.



Not surprisingly, these WGs submitted a multitude of recommendations based on their investigations. With respect to the condition of wiring on these older aircraft, typical findings, to mention just a few, included clamps being too small or too large, wires chaffing against structure, wires not adequately secured, wires tied to fuel lines, oxygen lines and hot air pipes. However, the most commonly found condition [and the one likely of most

concern] was the accumulation of dirt and debris consisting of lint, fluid spills and metal shavings; cleanliness would need to become paramount. The need for better wire marking and wire separation was noted. The difficulty of inspecting wires, particular due to inaccessibility and when in thick wire bundles, was also recognized.

In general, the need for a cultural change in the industry's and the regulator's approach to wiring was noted. During visual inspections, wires should no longer be considered as obstacles to be navigated around in order to perform other maintenance actions. A coffee spill does not just wet carpeting. As expected, there were training-related recommendations for more emphasis on wiring, recommendations for a standard format and standard minimum content in ESWPMs, and recommendations supporting the need for an Enhanced Zonal Analysis Procedure (EZAP).

ATSRAC WG Recommendations

It took these five ATSRAC WGs two years [Phase 1] to complete their taskings and submit their recommendations. In Phase 2, the ATSRAC, in light of the recommendations that were made, was tasked to recommend appropriate rulemaking action to effect implementation of the Phase 1 recommendations. Working Group 6 was formed to incorporate new wire system certification standards into Federal Aviation Regulation (FAR), Part 25 [Airworthiness Standards; Transport Category Airplanes]; revising this FAR would ensure that the ATSRAC recommendations addressing such matters as wire separation and markings would become a requirement for new aircraft designs. Guidance material for the development of Wire System Training Programs was produced that would be applicable to Air Carriers, Repair Stations and Training Institutions. Training would be tailored to each group's requirement; and there would be detailed procedures and precautions for hands-on mechanics and inspectors and general awareness information on the importance and criticality such as for flight and cabin crews and airline managers. The requirement to implement EZAP into an operator's approved maintenance program would be mandated by the FAA issuing a Special FAR. In addition, there would be regulatory action to mandate a one-time inspection of wiring systems within a five year time period. Although the FAA regulatory actions would only be applicable to US-registered aircraft, Transport Canada (TC) and European Joint Aviation Authority (EJAA) shared the FAA's concern and participated in ATSRAC activities; this was intended to allow TC and EJAA to mandate similar requirements simultaneously with the FAA.



Although the aging aircraft activities have been focused on large transport-category aircraft, it is recognized that small [less than 30 passengers and/or 7500 pound payload] transport-category aircraft are not immune to age-related deterioration of wiring. Accordingly, as part of the ATSRAC Phase 2 activities, WG 10 performed evaluations of 39 business-jet aircraft representing eight models by five manufacturers. The findings of WG 10 indicated that the wiring observations for both large and small transport-category aircraft were similar.

Mandating the initiatives mentioned above is a start to addressing wiring system concerns, but more activities need to be, and are being, done. A future issue of Feedback will describe the importance of EZAP and the other activities underway to address wiring concerns.

For those readers who wish more details about ATSRAC, a comprehensive description of the ATSRAC activities is posted on its web site www.mitrecaasd.org/atsrac/index.html.

Ray Raoux Manager, Corrective Action Continuing Airworthiness Aircraft Certification Transport Canada

Photos: Courtesy of J.R. (Rod) Digney

suspected unapproved PARTs



The submitters of the following Service Difficulty Reports (SDRs), received during the previous quarter, indicated that an unapproved part (SUP) was suspected. The list is provided here for information only and should not be construed as an identification of confirmed unapproved parts. In Canada, SUPs should be reported on a regular SDR form, indicating your suspicion of an unapproved part.

NOTE: Feed Back Issue 2/2003 contained errors in the Suspected Unapproved Parts. Below is the revised list that contains data related to Quarters 2 and 3.

MAKE/MODEL: AEROSPATIALE AS 350A

PART NAME: BEARING PART NUMBER: BOGUS PART CONDITION: SEIZED

SDR #: 20030319003

DESCRIPTION

THE DRIVE END BEARING SEIZED DURING FLIGHT AND SPUN IN THE BEARING LINER. THIS WORE THE OUTER DIAMETER OF THE BEARING RACE DOWN TO THE POINT THAT THE ARMATURE BEGAN RUBBING ON THE STATOR. THIS WEAR CONTINUED UNTIL SUFFICIENT TORQUE BUILT UP TO SHEAR THE DRIVE SHAFT AT THE QUILL SECTION. A TEARDOWN OF THE UNIT REVEALED THAT THE BEARINGS INSTALLED ARE BOGUS PARTS. THE CORRECT PART IS MADE BY THE OEM, APC, AND IS P/N 150SG1052B. THE BEARINGS FOUND WERE MARED "JAF 6203RS JAPAN". THE SEIZED BEARINGS SHED ITS SEALS, EXPOSING THE BALLS. IT IS OF DIFFERENT DESIGN COMPARED THE CORRECT PART. THE BOGUS BEARINGS HAVE TWO ROWS OF EIGHT BALLS, WHILE THE OEM PART HAS ONE ROW OF EIGHT BALLS. THIS UNIT WAS LAST OVERHAULED, ACCORDING TO THE COMPONENT HISTORY CARD, PRIOR TO MAY 1998. PER THE OWNER, THE UNIT WAS INSTALLED ON AN AIRCRAFT WHEN IT WAS IMPORTED, AND IT IS LIKELY THAT THE OVERHAUL WAS DONE IN PORTUGAL. THE BEARINGS ARE BEING HELD IN QUARANTINE PENDING TRANSPORT CANADA INSPECTION.

MAKE/MODEL: BOEING 737 217

SDR #: 20030321011

DESCRIPTION

THE UNSCHEDULED REMOVAL OF A B737 RUDDER PCU WHICH OCCURRED ON FEB 25/26 2003 IN YVR WAS IN RESPONSE TO A REQUEST FROM MATERIAL SERVICES BASED ON A FAA UNAPPROVED PARTS NOTIFICATION NO 2001-0089 DATED AUG 12,2002. THE SUSPECT PCU P/N 1U1150-3 S/N 737 WAS LISTED IN THE NOTIFICATION. -AIR CANADA SHOULD HAVE BEEN NOTIFIED OF THE SUSPECT UNITS BY TRANSPORT CANADA REF THE BAA CHAPTER 5 ITEM 510 511 513. -. THE REMOVAL OF THE RUDDER PCU P/N 1U1150-3 S/N 0737 DOES CROSS REFERENCE TO FTN ZKMZP8 AND DOES INDICATE THAT IT WAS INSTALLED ON A/C 511, BUT THE SEPTRE HISTORY INDICATES THE UNIT WAS REPAIRED ON MARCH 13, 2000. THIS INFORMATION WAS CONFIRMED IN ARTOS. THE FTN HISTORY IN ARTOS INDICATES THAT THE UNIT WAS INSTALLED ON A/C 511 ON MARCH 21, 2000 UNTIL THE PRESENT DATE. THE FAA NOTICE INDICATES THAT THE SUSPECT UNITS WERE IMPROPERLY OVERHAULED DURING A PERIOD OF JANUARY 3-8 2001. -. THE REPAIR STATIONS AND VENDORS MENTIONED IN THE NOTIFICATION ARE NOT ON OUR APPROVED LIST. ALSO OUR SHOP REPORT ESTABLISHED THAT THE PART HAD MFG CODES WE ARE PRESUMING THAT THE PART LISTED ON THE FAA LISTING HAS BE FORGED.

MAKE/MODEL: CESSNA 150

PART NAME: ALTERNATOR PART NUMBER: DOFF10300F PART CONDITION: FAILED

SDR #: 220030120006

DESCRIPTION

ON OVERHAUL OF THE ALTERNATOR THE RECTIFIER P/N ES4113, BRUSH P/N E4118, AND BEARING P/N ES4104. ALL THESE PART NUMBERS ARE NON-APPROVED. ALTERNATOR BUILD UP CARRIED OUT WITH APPROVED PARTS.

MAKE/MODEL: DE HAVILLAND DHC-2MK1

PART NAME: ELEVATOR PART NUMBER: C2T29A PART CONDITION: FAILED

SDR #: 20030131001

DESCRIPTION

PART WAS ACTUALLY REMOVED FROM STOCK FOR INSTALLATION BUT WOULD NOT FIT.

MAKE/MODEL: EUROCOPTER BO105 S CDN BS

PART NAME: BOLT PART NUMBER: LN93556X32 PART CONDITION: TOO LONG

SDR #: 20030605012

DESCRIPTION

DURING INSPECTION TO COMPLY WITH AIRCRAFT SERVICES CAMPAIGN NOTICE 351-73-20-12 THE BOLT CONNECTING THE N2 CONTROL CABLES TO THE N2 LEVER ASSEMBLY WAS FOUND TO BE TOO LONG. THE EXTRA LENGTH HAD BEEN COMPENSATED FOR WITH THE USE OF EXTRA WASHERS UNDER THE BOLT HEAD. THE INCORRECT BOLT WAS REMOVED AND CORRECT BOLT P/N LN9355-6X32 INSTALLED AND TORQUE SEAL APPLIED TO THE END OF THE BOLT TO NUT-PLATE CONNECTION.

FAA unapproved PARTs notification

No. 2001-00274 issued June 17, 2001

AFFECTED PARTS

Turbochargers.

PURPOSE

The purpose of this notification is to advise all aircraft owners, operators, manufacturers, maintenance organizations, parts suppliers, and distributors regarding improper maintenance performed on Garrett and RayJay/Roto-Master turbochargers.

BACKGROUND

A joint suspected unapproved parts investigation conducted by the Federal Aviation Administration (FAA), Department of Transportation - Office of Inspector General , and the Federal Bureau of Investigation revealed that Thunderbird Accessories, Inc. (Thunderbird), improperly maintained and approved for return to service Garrett and Raylay/Roto-Master turbochargers. Thunderbird, located at 5406 N. Rockwell, Bethany, OK 73008, previously held Air Agency Certificate No. IC2R893K.

Evidence revealed that Thunderbird had failed to accomplish maintenance in accordance with the manufacturers' maintenance manuals; Instructions for Continued Airworthiness; or other methods, techniques, and practices acceptable to the FAA. Discrepancies noted included stop-drilling beyond allowable limits on exhaust housings, and the installation of unapproved bolts and bearings that were not designed for extreme heat applications. The FAA has been unable to determine all models and serial numbers affected; therefore, all turbochargers approved for return to service by Thunderbird should be considered suspect.

RECOMMENDATIONS

Regulations require that type-certificated products conform to their type design. Aircraft owners, operators, maintenance organizations, parts suppliers, and distributors should inspect their aircraft, aircraft records, and/or parts inventories for turbochargers approved for return to service by Thunderbird. Suspect turbochargers installed on aircraft should be inspected for conformity to type design. If any are found in existing stock, it is recommended that the turbochargers be quarantined to prevent installation until a determination can be made regarding each part's eligibility for installation.

FURTHER INFORMATION

Further information concerning this investigation and guidance regarding the above-referenced turbochargers may be obtained from the FAA Flight Standards District Office (FSDO) given below. In addition to the above recommendations, the FAA would appreciate any information concerning the discovery of the turbochargers, the means used to identify the source, and the action taken to remove any turbocharger from service.

This notice originated from the Oklahoma City FSDO, 1300 S. Meridian Blvd., Suite 601, Oklahoma City, OK 73108, telephone (405) 951-4200, fax (405) 951-4282; and was published through the FAA Suspected Unapproved Parts Program Office, AVR-20, telephone (703) 668-3720, fax (703) 481-3002, web: http://www1.faa.gov/avr/sups/upn.cfm.

feedback Issue 3/2003 feedback

AME SYMPOSIA / TRADE SHOWS / WORKSHOPS 2003 - 2004

ONTARIO October 22 - 24

International Plaza Hotel & Conference Centre, 655 Dixon Road, Toronto ON M9W 1J4

Tel: 1-800-668-3656 or (416) 244-1711 Fax: (416) 244-8031

Internet: www.internationalplaza.com

NORTH WESTERN ONTARIO November 6 - 7

Victoria Inn & Conference Centre 555, Arthur Street West, Thunder Bay, ON Tel: 1-800-387-3331 or (807) 577-8481 Fax: (807) 475-8961 Internet: www.vicinn.com

QUÉBEC November 18 - 21

Hilton de l'Aéroport - Dorval 12505 Côte-de-Liesse, Montréal (Québec) H9P 1B7 **Tel** :1-800-567-2411 (514) 631-2411 **Fax**: (514) 631-01 **Internet**: www.hilton.com

PACIFIC January 28 - February 2

Best Western Richmond Hotel & Convention Centre 7551 Westminster Highway, Richmond, BC V6X 1A3

Tel: 1-800-663-0299 or (604) 273-7878 Fax: (604) 278-0188

Internet: www.richmond-hotel.ca

CENTRAL March 3 -5

Best Western Victoria Inn (Winnipeg Airport)
1808 Wellington Avenue, Winnipeg, MB R3H 0G3

Tel: 1-800-928-4067 or (204) 786-4801 Fax: (204) 786-1329
Internet: www.vicinn.com

WESTERN March 24 - 26

Coast Plaza Hotel & Conference Centre 1316 - 33rd Street NE, Calgary, AB T2A 6B6 Tel: 1-800-661-1464 or (403) 248-8888 Fax: (403) 248-9 Internet: www.info@calgaryplaza.com

ATLANTIC April 29 - May 1

CASINO Nova Scotia Hotel
1919 Upper Water Street, Halifax, Nova Scotia B3J 3J5
Tel: 1-866-425-4329 or (902) 421-1700 Fax: (902) 422-5801
Internet: www.casinonovascotia.com

service difficulty reports

Received by Transport Canada from 1 April 2003 to 30 June 2003

MAKE/MODEL	ATA	PART NAME	PART NO. PA	ART CONDITION	SDR NO.	RGN
aircraft						
AEROSPATIALE						
AS 350B AS 350B AS 350BA AS 350BA	2913 6730 2910 6300	HYDRAULIC PUMP B HYDRAULIC SERVO HYDRAULIC PUMP B BEARING LOCK	704A33690004 SC5083 704A33690004 350A37118322	CRACKED FAILED CRACKED/TORN BROKEN	20030409001 20030527012 3 SDRs 20030429002	QUE ONT
AS 350BA AS 350B2 AS 350B2 AS 350B2 AS 350B2	7250 3213 6720 6730 7931	MO2/MO3 BOLT HYDRAULIC HOSE SERVO, HYDRAULIC ENGINE OIL PRESS	70BM02106003109 22201TK080015X 704A34412047		20030526012 20030530013 20030409010 20030527008 20030612004	PNR PNR ONT
AS 350D AS 355F1 AS 355F2 AGUSTA	6300 5531 5302	BEARING LOCK RIB BULKHEAD FRAME	350A37118322 350A14111925 350A23103600	BROKEN CRACKED CRACKED	20030502005 20030617002 20030407010	ONT QUE
A109 All AIRBUS	5302	VERTICAL FIN BUL	1090372441	CRACKED	20030527003	PAC
A310 308 A310 308 A310 308 A319 112 A319 114 A319 114	2900 3230 5240 3242 5520 5610	HOSE ASSEMBLY HYDRAULIC HOSE OIL ACCESS DOOR BRAKE ASSY R/H ELEVATOR WINDOW	AE7060322 AE7060322 2242277501 21600 NP1653114	DAMAGE RUPTURED BROKEN FIRE	20030403001 20030421001 20030522002 20030521005 20030530004 20030505002	QUE QUE QUE
A319 114 A320 211 A320 211 A320 211 A320 211 A320 212 A340 313	5610 5610 2720 2742 2780 5610 2133 7830	WINDOW WINDSHIELD RUDDER POS.FEEDBD STAB ACTUATOR (STEADY BEARING WINDSHIELD OUTFLOW VALVE THRUST REVERSER	NP1653114	SHATTERED DAMAGED FAILED	20030504002 20030527004 20030410001 20030526005 20030624002 20030523003 20030409007 20030611003	QUE QUE QUE QUE QUE
A100	3233	ACTUATOR (ALL) (F	99810057652	STRIPPED	20030604003	PNR

ACTUATOR ACTUATOR/VALVE

ENGINE TEMP IND PLATE, EXHAUST S

FLAP ACTUATOR

BLOWER AAAY

WINDSHIELD

NOZZLE FLAP DYNAMIC BRE

FUEL TUBE

ANTI-ICE DUCT

PNEUMATIC AREGUL

BALANCE RESISTOR

PRESSURE SWITCH BUSHING

ELEVATOR CABLE

HINGE, ELEVATOR

90 DEGREE DRIVE

DIRECTIONAL GYRO

ANTI-ICE FLEX TUBE

ANTI-ICE FLEX TUBE

ANTI-ICE FLEX TUBE

CIRCUIT BREAKER

RUDDER RETURN

PRINTED CIRCUIT UPPER TORQUE

L/H CABIN WINDOW

ACTUATOR

MOTOR

CONTROLLER

LIGHT SOCKET ASS

INTERCOSTAL BULK

NOSE GEAR ACTUATOR 11238002221

BOLT

INSERT REINFORCE SPLINE TRAILING

2160

5510

5520

7720

0000

2711

2752

3260

3600

7797

2752

2820

3020

3422

7160

7160 7510

5620

3230

3230

	704A33690004 SC5083 704A33690004 350A37118322 708002106003109 22201TK080015X 704A34412047	CRACKED FAILED CRACKED/TORN BROKEN UNSERVICEABLE U/S CRACKED	20030409001 20030527012 3 SDRs 20030429002 20030526012 20030530013 20030409010 20030527008	QUE PNR QUE ONT PNR PNR PNR ONT
	350A37118322 350A14111925 350A23103600	BROKEN CRACKED CRACKED	20030612004 20030502005 20030617002 20030407010	PNR ONT QUE ONT
	1090372441	CRACKED	20030527003	PAC
	AE7060322 AE7060322 2242277501 21600	DAMAGE RUPTURED BROKEN FIRE	20030403001 20030421001 20030522002 20030521005 20030530004	QUE QUE QUE QUE QUE
	NP1653114 NP1653114 2727000601800 47145131 8018170001 NP1653115 90231570381	SHATTERED DAMAGED FAILED	20030530004 20030505002 20030527004 20030410001 20030526005 20030624002 20030523003 20030409007 20030611003	QUE QUE QUE QUE QUE QUE QUE QUE QUE
	99810057652 NYLC9871 KNKDL428 115610010125	STRIPPED NORMAL STRIPPED CRACK	20030604003 20030428004 20030414002 20030408006 20030428006	PNR ONT PNR PNR ONT
R	1019100751 130909B38 1295210511 11238002221 1147512 501A 1013841761 570383 50160020 10138402522 H11D59 SM60D7 5052443929 5052442034 503801536 50921620235 50921620235 9091010017 6226163002 9091009913 9091009913 9091009913 7091009913 9091009913 9091009913 9091009913	LOOSE & LEAKING FAILED LEAKING FAILED LOOSE CONNECTION BAD BEARING FAILED U/S CRACKED LEAKING INTERMITTENT INTERFERANCE CRACKED/CORRODED O.K. GROOVED CRACKED PARTIAL FAILURE CRACKED CRACKED CRACKED CRACKED CRACKED BROKEN BROKEN GOOD	20030627010 20030610005 20030610007 20030424006 20030425005 20030425005 20030425005 20030425005 200304004 20030604004 20030604004 20030424007 20030512001 20030416002	ONT PNR PNR PNR PNR PNR PNR PNR PNR ONT PAC ATLE QUE ONT ONT ONT ONT ONT ONT ONT ONT ONT ONT
	504100491 MS24171D1	CRACKED NO FUNCTION	20030514004 20030619002 20030516002	PNR PAC ONT
	KNE1018100329	SEPARATED CRACKED	20030603002 20030610009	ONT ONT
	9981005765 MC815AS1 1153800025	HARD TO TURN STUCK IN UP MODE POPPED C B	20030508003 20030501001 20030501008	ONT ONT ONT

A.Printerior		route				
MAKE/MODEL	ATA	PART NAME	PART NO.	PART CONDITION	CTRL NO.	RGN
200 200	3320 5210	WIRE HARNESS W18 PASSENGER/CREW D	L139A22L139B22	BURNT	20030508002 20030612001	ONT ATL
200 200 200	5520 5610	TORQUE TUBE-ELEV WINDSHIELD	1016100191	CORROSION FRACTURED	2 SDRs 20030529002	ATL QUE
200 95B55	7720 2720	FIRE DETECTOR RUDDER PEDAL	302158 505243263	SERVICEABLE BROKEN/WORN	20030323002 20030429001 2 SDRs	ONT VAR
95B55 99A	2810 3233	AIRFRAME END CAP	4377113	CRACKED	20030517001 20030424002	PNR ONT
BELL						
204B 204B	5210 5302	UPPER LONGERON SPARE	205030619 20403082506	NEW CRACKED	20030421005 20030610011	ONT PNR
204B	5313	LONGERON	204032806079A	ORIGINAL	20030421004	ONT
205A 1 205A 1	2910 6520	HYDRAULIC CHECK CONTROL TUBE ASSY	2120763643 209011710005	CRACKED CRACKED	20030404003 20030610003	PNR PNR
206A	5532	TAIL FIN ASSEMBLY	206020113005	CRACKED	20030409013	PNR
206B	0000	GOVERNOR		FAILED	20030627005	PNR
206B 206B	3213 5347	SADDLE ANGLE	D26651 206031103043	CRACKED CRACKED	20030407006 20030528004	PAC ONT
206B	5511	SPAR	206020120011	CRACKED	20030526011	PAC
206B	6320	DRAG PIN ASSY	2060315095	GOOD	20030617003	PNR
206L	5302	FITTING UPPER	LE206031329003	CRACKED	20030617001	ONT
206L 1 206L 1	5302 6320	FRAME UPPER CASE	206032308103S 206040532005	CRACKED CRACKED	20030528005 20030528006	ONT
206L 3	0000	ENGINE MOUNT LEG	206069105101	CRACKED	20030320000	PAC
206L 3	6320	TOP CASE	206040532005	BROKEN	20030401005	PNR
212 212	2912 5302	ELBOW BULKHEAD	MS219266 212030156047	CRACKED CRACKED	20030625001 20030603010	ONT PNR
212	5310	ANGLE	205030712037	CRACKED	20030603010	PNR
212	6320	BEARING	205040249003	METAL SEPARATION	20030425003	PAC
212 212	6320 6330	SUPPORT CASE LIFT LINK	212040054007	BROKEN	20030613009	PAC PAC
212	6420	MAIN ROTOR BLADE	212030104101 212015501115	BROKEN CRACKED	20030530001 20030530003	QUE
212	7310	BELL CRANK	212060745001	FAILED	20030416003	ONT
407 412CE	6220	STUDS (4)	407010105101	CRACKED	20030523001	PAC PNR
412CF BELLANCA	6230	BEARING & LINER	204010433001	LOOSE	20030401006	PINK
8GCBC	3220	FRAME TUBE	7147061	BROKEN OFF	20030616003	ONT
8GCBC# BOEING	2752	LH UPPER FLAP	CA19023	FRAYED	20030616002	PNR
727 217	2612	CANNON PLUG	BACC45ZT18A31P6	SHORTED	20030411001	ONT
727 217 727 22C	0000	SLAT ACTUATOR			20030627018	PAC
727 223 727 223	0000	HYD FLEX LINE	BACH8B04NN0157 6572719233	RUPTURED DAMAGED	20030627012	ONT
727 225	0000	WING TIP ASSEMBLY VERT STAB LEADIN CABLE ASSY WIRING SLAT EXT. FUSELAGE	03727 19233	FAILED	20030627013 20030627009 20030627001 20030520003	ONT
727 225	2710	CABLE ASSY	BACC13AP6E285	8 CHAFED	20030627001	ONT
737 2Q9 737 2Q9	2797 5330	WIRING SLAT		BURNT ODOR	20030520003	ATL ATL
737 2Q9 737 2Q9	5330	EXTERIOR FUSELAGE		CRACKED CRACKED	20030409017	ATL
737 2T2C	3230	LANDING GEAR RET			20030409018 20030528003	PNR
737 2T5 737 2T7	2900 5610	AILERON PCU WINDOW	65447619		20030417011 20030505003	PAC PAC
737 217	5330	L/H SIDE OF FUSE	65C343482	CRACKED	20030505003	PNR
737 217	2600	APU			20030620004	PAC
737 242C 737 275C	0000 3260	NIL/UNKNOWN LANDING GEAR POS			20030620006 20030418001	PNR PNR
737 281	2710	AILERON PCU	654476114	CAP BROKEN	20030418001	PNR
737 76N	0000	MAIN WHEEL ASSEMBLY			20030618001	PNR
737 76N 767 333	2897	WIRE	NA 1016E630E	N/A	20030425010	PNR
BOMBARDIER	2565	CHUTE	101656305		20030528013	QUE
BD 700 1A10	2731	ELEVATOR TAB CON			20030520004	NCR
CL600 2D24 BRITISH AEROSPA	2130	NIL/UNKNOWN			20030426001	NCR
BAE 146 200	2621	DISCHARGE HEAD	3473641	CONTAMINATED	20030514015	ATL
BAE 146 200 BAE 146 200	5310	UPPER WHEELWELL	NA	CRACKED	3 SDRs	ATL
BAE 146 200 HS 748 2A	5610	FLIGHT COMPARTMENT HOSE ASSEMBLY		DDOVEN	20030403003	ATL
HS 748 2A HS 748 2A	2110 2710	OPERATING LEVER	139379 1F11792	BROKEN FRACTURED	20030611001 20030424001	QUE ONT
HS 748 2A	5200	LOCK SOLENIOD	UK	OVER HEATED	20030407004	ONT
CANADAIR CL215 1A10	0000	CYLINDER		CRACKED	20030623001	DNID
CLZ IS IA IU	0000	CILINDER		CRACKED	20030023001	LINK

B100

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C90

C90A C90A C90A C90A C90A

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feedback

MAKE/MODEL ATA	PART NAME	PART NO.	PART CONDITION	SDR NO.	RGN	MAKE/MODEL	ATA	PART NAME	PART NO. P	ART CONDITION	CTRL NO. RGN
CL215 1A10 0000 CL2156B11(CL215T) 5720 CL600 1A11(600) 3010 CL600 2B16(604) 2910 CL600 2B16(604) 3246	REAR TRUNNION DOUBLER PICCOLO TUBE NIL/UNKNOWN JANTE ROUE DE NE	1606042 215K10105154 600800398 6047523865	WORN CRACK MISLABELED CHAFED	20030506005 20030407009 20030606006 20030401007 20030409003	QUE ONT ONT NCR	208B 310L 310L 310R 337A	3221 3230 3232 8000 2782	BUSHING DOWNLOCK SPRING GEAR DOORACTUATOR STARTER CONTACT FLAP CONTROL CABLE	008504691	MISSING BROKEN WELD CRACKED ARCED	20030430007 PNR 20030520005 ONT 20030506001 ONT 20030605003 PNR 20030606010 PNR
CL600 2B19(RJ) 0000 CL600 2B19(RJ) 0000 CL600 2B19(RJ) 0000 CL600 2B19(RJ) 2120 CL600 2B19(RJ) 2121 CL600 2B19(RJ) 2215	AIRCRAFT PSEU SEE TEXT BULKHEAD CHECK FLAPPER VALVE CAPSTAN DRIVE	V92E204 101270 GE6288713001	FAIL CORROSION DISMANTLED GROOVED DOWELS SHEARED	20030623002 20030619004 20030624001 20030406001 20030404009 20030603013	QUE QUE NCR ATL	337G 402B 402B 414 500 525	2752 3230 3244 2425 3234 0000	CABLE ASSEMBLY ADJUSTING FORK MAIN GEAR TIRE ALTERNATOR LANDING GEAR STARTER GENERATOR	14601007 50452112 0773450 ALV9407 66710	SHEARED BLOWN FAILED	20030602001 ATL 20030523005 PAC 20030521008 PAC 20030515005 PNR 20030408001 ONT 20030626004 QUE
CL600 2B19(RJ) 2421 CL600 2B19(RJ) 2710 CL600 2B19(RJ) 2710 CL600 2B19(RJ) 2761	INTEGRATED DRIVE AILERON CONTROL PCU OUTPUT LINK POWER CONTROL UNI	755469B 272955 Γ 270007	METAL IN FILTER WORN	20030603012 20030516001 20030514002 20030612009	QUE QUE NCR QUE	550 550 550 550	0000 0000 0000 2750	ESCAPE HATCH SEAL INSULATION ISOLATION PISTON SWITCH ASSEMBLY	651125619 66422172 656517543	STUCK SCORCHED PUNCTURED MIS-RIGGED	20030626001 ONT 20030627002 PAC 20030625004 ONT 20030605014 ONT
CL600 2B19(RJ) 2910 CL600 2B19(RJ) 2910 CL600 2B19(RJ) 2916 CL600 2B19(RJ) 3234 CL600 2B19(RJ) 3250	HYDRAULIC LINE NIL/UNKNOWN HYDRAULIC RESERVE GEAR SELECTOR ELECT HYD SHUT	6017523843 VA750005000 O16480101	CHAFED LEAKING DEFECTIVE	20030419001 20030515006 20030501003 20030421006 20030603001	NCR ATL QUE	550 550 550 560 560XL	3242 5753 7220 7830 3244	STATOR DISK RIB EXTENSIONS HEATED LIP PICO HYDRAULIC HOSE TIRE MICHELIN	1338932 6525115 AE1009195G0130 Al0316138	BROKEN FRETTING WEAR CRACKED RUPTURED EXTERNAL O.K	20030603006 PAC 20030610001 ONT 20030409011 PNR 20030604002 PNR 20030528012 QUE
CL600 2B19(RJ) 3250 CL600 2B19(RJ) 3418 CL600 2B19(RJ) 4900 CL600 2B19(RJ) 5320	VALVE AOA VANE APU ISOLATION BLANKET	750006000 601R591541 601R404672381	FAILED OUT OF TOLERANCE FAILED	20030501005 20030609010 20030514017 20030514011	QUE QUE QUE	650 CONVAIR 340 340	3260 3230 5314	WIRE VALVE KEEL BEAM	KF580532000012	BROKEN	20030520001 QUE 20030620005 PAC 20030404005 PAC
CL600 2B19(RJ) 5340 CL600 2B19(RJ) 5610	NIL/UNKNOWN COCKPITSIDE WINDSHIELD	601R3303312	CRACKED	20030417002 20030612008	QUE •	CURTISS WRIGHT	1000	COTTER PIN	MS24665143	NEW	20030528015 PNR
CL600 2B19(RJ) 5610 CL600 2B19(RJ) 5610 CL600 2B19(RJ) 5610 CL600 2B19(RJ) 5610 CL600 2B19(RJ) 5700	COCKPIT WINDSHIELD COCKPIT WINDSHIELD L/H SIDE WINDOW SIDE WINDOW LEADING EDGE	601R3303313 601R3303311 NP1393221 601R120445	SHATTERED SHATTERED CRACKED	20030507001 20030514001 20030530011 20030505001 20030419002	NCR QUE NCR	DASSAULT FALCON 20D FALCON 900 FALCON 900 DEHAVILLAND	7931 3411 7830	OIL PRESSURE TRAF MOISTURE TRAP SEAL/SHIELD	11346AA 900000 FGFB583800308	CRACKED CRACKED	20030612002 ONT 20030514006 PNR 20030428008 ONT
CL600 2B19(RJ) 5740 CL600 2B19(RJ) 5741 CL600 2B19(RJ) 5753 CL600 2B19(RJ) 7200	BUSHING BUSHING ENGINE TURBINE	601R14501-1 60	MIGRATING MIGRATION	20030417004 6 SDRs 20030403004 20030515007	QUE QUE NCR QUE	CS2F 2 DHC 2 MKI DHC 2 MKI DHC 2 MKI	3260 2720 5511 5523	DWNLCKMICROSWITCH MOUNT ASSY COPILOT FRONT SPAR TORQUE TUBE ASSY	C2CF1777A CFTP57 C2T29A	LOOSE CRACKED BROKEN	20030521009 PNR 20030613004 PAC 2 SDRs VAR 20030609009 ONT
CL600 2B19(RJ) 7603 CL600 2B19(RJ) 7603 CL600 2B19(RJ) 7897 CL600 2C10(RJ) 2613 CL600 2C10(RJ) 5610	THROTTLE CONTROL THROTTLE GEAR BOX STOW SWITCH BLEED SENSING LO SIDE WINDOW	22850741115 GG670801021 NP1393225	MISADJUSTED CRACKED	20030518001 20030518002 20030514018 20030530014 20030406002	QUE NCR QUE	DHC 2 MKI DHC 3 DHC 3 DHC 6 300 DHC 6 300	7400 2742 7110 2731 5534	PLUG ACTUATOR AS ABOVE LEVER CONTROL SPAR, FORWARD	SKL32138AN C3CF2907 AC35500 ASC6CF14191 C6TP103927	SHORTED FAILED CRACKED FAILED CRACKED	20030526009 PAC 20030416004 ATL 20030620003 ONT 20030527011 PNR 2 SDRs ONT
CESSNA A185E 5551 A185F 2750 A188B 2701	R/H AFT SUPPORT FLAP PULLEY SPRING	07123022 S3782 16001041	CRACKED WRONG PART BROKEN	20030501002 20030521006 20030611005	ONT QUE ATL	DHC 6 300 DHC 6 300 DHC 6 300 DHC 7 102	5534 5753 7930 2612	SPAR, AFT RH INBOARD FLAP CHIP DETECTOR WIRE	C6TP103931 C6WF10024 3034081 261030013A22	CRACKED LOOSE RIVETS CRACKED CHAFED/BROKEN	20030505009 ONT 20030613005 ONT 20030609003 ONT 20030514008 ONT
TU206G 2820 TU206G 5344 T310P 3230I U206B 8530	FUEL LINE LEFT DOOR POST DLER BELLCRANK VACUUM AIR PUMP	1200106253 TU206G 08421022 212CW	SEVERED CRACKED BROKEN FAILED	20030606004 20030514016 20030603007 20030528001	ONT PNR ONT	DHC 7 102 DHC 7 102 DHC 8 102 DHC 8 102	3234 5730 0000 0000	LGSV SOLENOID LIGHTNING SEAL BEARING NLG STRUT	574205A CVR67 MB543DDFS464 8800	FAILED SWOLLEN SEIZED WORN	20030616004 ONT 20030609002 ONT 20030630001 ATL 20030619007 ONT
U206C 5310 U206G 5330 120 2720 150G 2720 150L 5510	L/H FWD DOOR POS SKIN, LOWER CENT RUDDER BELLCRANK STOPNEW RIB ASSY LEADING	121340615 1211650 300433113 0432001646	CRACLED CORRODED CRACKED CRACKED BROKEN	20030410004 20030505006 20030515003 20030517002 20030423006	ONT ONT ONT	DHC 8 102 DHC 8 102 DHC 8 102 DHC 8 102 DHC 8 102	2750 2761 2761 2900 2900	TORQUE SENSOR BR LIFT DUMP HYDRAU UNLOADING VALVE HYDRAULIC LINE SHUTTLE VALVE	85711423101 82950010269 659603 82960010115 50841	BROKEN & CRACKED LEAKING CRACKED BROKEN	20030530005 PNR 20030501004 ATL 20030404008 ATL 20030620001 QUE 20030602010 ATL
150L 5711 150M 0000 150M 5510 152 8530	REINFORCEMENT CYLINDERS RIB ASSY LEADING ENGINE	043200115 0432001646	CRACKED BROKEN FAILED	20030401004 20030626009 20030423004 20030609012	ONT ONT ONT OUE	DHC 8 102 DHC 8 102 DHC 8 102 DHC 8 102	2910 3232 3242 3260	FLIGHT SPOILER CYLINDER ASSY BRAKE ASSY PSEU ILD. GEAR C	H82950010369 82970019007 214665 841005	PIN HOLE CORRODED CARD FAULT	2003052002 ATL 20030520002 ATL 20030513004 PNR 20030423003 ATL 2 SDRs ATL
172M 0000 172M 2430 172M 2434 172M 7160	FUEL LINE ASSY CIRCUIT BREAKER ALTERNATOR DOFF CARB HEAT CONTRO	05001183 S159060L 10300J S123017	CHAFED FAILED FAILED BROKEN WIRE	20030627015 20030423002 20030425006 20030430005	ATL ATL ONT PNR	DHC 8 102 DHC 8 300 DHC 8 301 DHC 8 311	5300 3220 3230 2900	PROX SENSOR EDP HYDRAULIC	4426X210 8600113 858601 82970010477	FAULT CODE PIN HOLE	20030502001 ATL 20030625008 NCR 20030618003 ATL 20030507002 ATL
172N 3244 172N 5712 172P 5541 172R 2562 172R 8011 172RG 2311 172RG 5343	5.00 * 5 TIRE RIB-LH RUDDER TOP RIB LANYARD, ELT STARTER PM AUDIO PANEL FITTING	Al53716 05230301 05331531 05118204 2401H 24700032 24130023	BLISTER CRACKED CRACKED CHAFED FAILED CRACKED	20030407001 20030408005 20030410005 20030423007 20030528007 20030603005 20030409004	PNR QUE ONT ONT ONT	DHC 8 311 DHC 8 315 DHC 8 315 DHC 8 400 DHC 8 400 DHC 8 400 DHC 8 400	3246 2721 2913 2130 2610 3230 3260	WHEEL BEARINGS ROTARY SWITCH ENGINE DRIVEN CABIN INDICATION FIRE LOOP DETECT WOW2 / NOSE WHEE NLG DOWNLOCK PRO		SPALLED/PITTED INTERNAL FAULT	20030526008 ATL 20030625007 NCR 20030515002 NCR 20030624003 NCR 20030609001 NCR 20030624004 NCR 20030515001 NCR
182S 2500 182S 5320 182S 5320	CYLINDER LOCK BULKHEAD BULKHEAD	05142132 07137878 07137879	COLLAPSED CRACKED CRACKED	20030427001 2 SDRs 20030414006	ONT ONT ONT	DHC 8 400 DHC 8 400 DIAMOND	3297 3297	HARNESS WOW 2 / NOSE WHEE	471515 L		20030625002 NCR 20030625003 QUE
182S 5320 208B 2750 208B 2810	LH FUSELAGE SKIN FLAP MOTOR FUEL RESERVOIR	07133343 C145350 261601741	CRACKED UNKNOWN CRACKED	20030414004 20030421003 20030514010	PNR :	DA 20 A1 DA 20 A1 DA 20 A1	0000 2720 2720	ELEVATOR HINGE LOCOLOC OVAL OVAL SLEEVE	B2053840204 SLESL24P SL24P	BROKEN OFF	20030627017 ONT 20030411006 ONT 20030404010 ONT

MAKE/MODEL	ATA	PART NAME	PART NO.	PART	CONDITION	CTRL NO.	RGN
DOUGLAS							
DC9 83	2210	ELECTRICAL WIRE	2C610Y20RD		CHAFED	20030424008	
DC9 83 DC9 83	3425 5210	PRIMARY FLIGHT CABIN DOOR SWITCH	D4056510903 2028B		FAILED NEED CLEANING	20030514014 20030529004	
EMBRAER			20202		11223 0223 111110		
EMB 120ER EUROCOPTER DE	5320	COMPOSITE PANELS				20030528011	QUE :
BK117 B 1D	5302	SPLICE	1172213201		CRACKED	20030527014	PNR :
BK117 B 1D	5310	SPLICE	1172213202		CRACKED	20030527015	PNR :
BK117 B 1D BO105 C BS	6320 2910	U/K VALVE BODY MANIFOLD	0 133756E01		SCRAP	20030530012 20030516005	
BO105 C BS BO105 C CDN	6320	BEARING	4638302023		SCRAP	20030405001	ONT
BO105 C CDN BO105 C CDN	6200 6200	BOLT BOLT	LN93556X32 LN93556X32		LOSS OF TORQUE TOO SHORT	20030605013 6 SDRs	VAR
BO105 C CDN	6320	TAPERED ROLLER B	4619310069		SCRAP	20030405002	ONT :
BO105 C CDN	7321	SHAFT	UNK		SHEARED	20030527009	ONT
BO105 C CDN BO105 S CDN BS	7600 7240	BOLT, HEXAGON OUTER COMBUSTION	LN93556X32 6870992J		GOOD CRACKED	20030502003 20030404001	
EUROCOPTER FF	RANCE						•
EC 120 B FAIRCHILD	2400	LACU	911TS08Y		FAILED	20030409012	PNR :
SA227CC	7230	P3 TUBE	8943825		FRACTURED	20030410006	ONT :
GULFSTREAM G 1159A	0000	BALL RETURN TUBE	B120344		SHEARED	20030613001	ONT
G 1159A G 1159A	2910	HYDRAULIC TUBES	1159P20518		CHAFED	20030409008	ONT :
HUGHES							:
369D 369D	2820 5511	FUEL BYPASS SWIT SPAR	369H81443 369D23623		CRACKED CRACKED	20030411002 20030618007	
369D	6210	M/R BLADE PIN	369A10045		BROKEN	20030605010	PNR :
369D	6410 6710	TAIL ROTOR BLADE COLLECTIVE SUPPORT	369D21640501 369A7339		UNSERVICEABLE BROKEN	20030408003 20030425009	
369D LEARJET	0710	COLLECTIVE SUFFORT	309A1339		BRUKEN	20030423009	FINE .
35A	0000	KEYWAY LINER	9544008		BENT	20030618004	
35A 35A	2910 3246	TUBE ASSEMBLY WHEEL HALF	2607003453 95440263		HOLE CRACKED	20030605009 20030618002	QUE • PNR •
35A	7321	FUEL CONTROL UNIT				20030605015	QUE •
45 LOCKHEED	7532	HP BLEED VALVE	6621000000073	3	FAILED	20030414001	ONT :
L 1011 385 1 14	3510	L/P HOSE 02	4019374		LEAKING	20030428001	
L 1011 385 3 MITSUBISHI	7280	OIL CAP	LK58367			20030428003	QUE
MU 2B60(USA)	6140	TACHOMETER GENERA	A G34		SHEARED	20030505004	ONT :
MORAVAN 7242L	3250	NOSE WHEEL CABLE	Z4242390000		BROKEN	20030619006	ONT
PILATUS		NOSE WHILL CABLE	24242390000		BROKEN		•
PC 12 45 PC 12 45	0000	LED DISPLAY	9755084910			20030630002	
PC 12 45 PC 12 45	2750 2822	FLAP FLEXIBLE DR BOOST PUMP	9688411404			20030516006 20030508001	
PC 12 45	2897	BOOST PUMP	9688411404		. =	20030422001	ONT
PC 12 45 PC 12 45	2932 3150	LANDING GEAR HYD WARN LIGHT PANEL	9738114306 9728132012		LEAKING	20030509001 20030507003	
PC 12 45	3242	ROTOR DISK	23203500		BROKEN	20030407005	PNR :
PC 12 45 PC 12 45	3246 3260	FAIRING PROXIMITY SWITCH	15702100		CRACKED	20030430008 20030410007	
PC 12 45 PC 12 45	3340	LANDING LIGHT AS	973303311 9728769401		REPAIRED	20030528016	PAC
PC 12 45	3411	STATIC LINE	0754404404		FAILED	20030527010	
PC 12 45 PC 12 45	3418 3497	AOA SWITCH TORQUE TRANSMITTER	9754421421 K22A24		SHORTED OUT	20030428005 20030527016	ONT PAC
PC 12 45	7421	PLUG , IGNITOR	311614401			20030430009	PNR :
PC 12 45 PIPER	7712	TORQUE LIMITER	311629401		MALFUNCTIONING	20030403002	ONT •
PA18 150	3220	BOLT	AN721		SHEARED	20030409006	
PA18 150	5730 3210	WING STITCHING	1754400		BROKEN	20030529003	
PA23 250 PA23 250	3210	DRAGLINK FITTING TUBE	1754400 1742002		CRACKED CRACKED	20030401002 20030401001	
PA23 250	3230	DOWN LOCK MECHAN			HARD TO MANOEUVRE	20030409015	QUE :
PA28 160 PA30	2710 0000	AILRN BELCRANK LUG - TRANSMISSI	BR6210200 2176000		CRACKED BROKEN	20030409014 20030627016	
PA30	3230	MAIN GEAR CONDUI	455180		FAILED	20030610004	PNR •
PA31 PA31	2210 2730	CLAMPS PITCH SERVO BRDL			JAMMED JAMMED	20030606002 20030606003	
PA31	2750	SHAFT FLEX DRIVE	486597		CRACKED	20030501009	PNR :
PA31 350	2121	ELECTRIC MOTOR			BURNT ODOR	20030609011	QUE •
PA31 350	2730	AFT. TRIM CABLE	41734078		BROKEN	20030505005	AIL

MAKE/MODEL	ATA F	PART NAME	PART NO.	PART CONDITION	CTRL NO. RGN
PA31 350 PA31T PA31T PA31T PA38 112 PA44 180 PA44 180	3244 2731 3210 3250 7332 2300 2710 3220	NOSE TIRE & TUBE BEVEL GEAR PIVOT ARM STEERING ARM BOL FUEL PRESSURE RADIO BALANCE CABLE BOLT	6006 4536700 42042000 AN3H7A TR486439 62701123 NAS464P427	TWO TEETH BI CRACKED SHEARED FAILED FAILED FRAYED SHEARED	20030613003 ATL ROKEN 20030421007 ONT 20030619005 QUE 20030609008 ONT 20030525006 ONT 20030528009 ONT 20030528009 ONT 20030404002 ONT
ROBINSON R44	6300	ACTUATOR	C0511	FAILED	20030514003 PNR
R44 SCHWEIZER	7314	FUEL PUMP			20030527007 PNR
269C 269C	6200 6200	BEARING PULLEY SHAFT	269A505085 269A54985	FAILED UNDERSIZE	20030425008 ONT 20030425007 ONT
SHORT&HARLAN SD3 60 SD3 60 SD3 60 SD3 60	0000 0000 0000 7600	CIRCUIT BREAKER, SPRING SWITCH, MASTER POWER CONTROL	SM600BA 500221XXX321 LHA4B6101 CAA552326	FAILED CLOSE SPRING BROK JAMMED BROKEN	
SIKORSKY S61L S61N S61N S76 SWEARINGEN	5302 6410 6710 4130	LOWER FITTING TAIL ROTOR BLADE COLLECTIVE FRICT SEAL, OUTPUT	S61206612000 S611730101049 SB8208102		20030421008 PAC 20030610002 PAC 20030415003 ATL 20030416005 PAC
SA226TC SA226TC VICTORY AIRCRA	0000 2110 AFT	RECEPTICAL TUBE ASSY	27200637 2784142007	CRACKED BROKEN	20030620002 PNR 20030527005 PNR
AVRO LANCASTER MK)	X3240	SAME	SAME	STRETCHED	20030615001 ONT

engines

ALLISON						
250	7230	BEARING	23004538		20030425004	
250-C20	2800	ACCUMULATER	6875224	GOOD	20030421009	PAC
250-C20	8012	COMPRESSOR CASE	2305714	GOOD	20030612007	
250-C20B	7321	FUEL CONTROL UNIT	23065104	FAILED	20030430006	
250-C20B	7323	DRIVE GEAR		FAILED	20030527013	
250-C20R/2	7240	OUTER COMBUSTION	23034114	CRACKED	2 SDRs	PNR
250-C30	7931	CVR ASSY, GB,	PW23059576	UNSERVICEABLE	20030509002	
250-C30S	7260	OIL PUMP	23031459	UNKNOWN	20030415004	
250-C30S VCO LYCOMING	7261	HYDRODYNAMIC OIL	23063371	DISINTEGRATED	20030409002	QUE
AEIO-360-A1B6		CAMSHAFT	LW18840	SPALLED	20030626007	PAC
AEIO-360-A1B6	7810	INTERNAL BAFFLE	UK	BROKEN	20030526007	ONT
AEIO-360-A1B6		CAMSHAFTL	W18840	SPALLED	3 SDRs	PAC
GO-480-G2D6	0000	EXHAUST VALVE	71381	BROKEN	20030627014	
IO-360-A1B6D	8530	VALVE	7 130 1	BROKEN	20030604001	
IO-540-K1B5	8520	ENGINE		FAILED	20030605004	
LTIO-540-J2BD	8520	CRANK CASE		CRACKED	20030523004	
LTS-101-600A-3		FUEL CONTROL UNI	430128807	FAILED	20030530010	
O-235-L2C	7414	MAGNETO	4381	U/S	20030526006	ATL
O-320-B2C	7921	ENGINE OIL COOLER	20008A	LEAKING	20030603008	PNR
O-320-E2D	7300	ENGINE		FAILED	20030530009	
O-320-E2D	7322	CARBURETOR	10S135	BROKEN	20030604007	
O-320-E2D	7414	MAGNETO	4371	FAILED	20030502002	
O-320-H2AD	7414	BREAKER POINTS	ES10384585	WORN	20030514007	
O-320-H2AD	8520	CRANKCASE	LW15272	CRACKED	20030605006	
O-360-A1F6D	7414	MAGNETO	ES1068255511	FAILED	20030424005	
O-540-F1B5	8530	ENGINE			20030407007	
TIO-540-AJ1A	8520	CLAMP	40D22382	DEFECTIVE	20030530002	
TIO-540-A2B	6122	CRANKSHAFT	79027	FAILED	20030606008	
TIO-540-A2B	8520	CRANKCASE	11F20022S3	LEAKING	20030514013	
TIO-540-A2C	7310 8500	TUBE ASSEMBLY	LW120980100 75414	CASSIE	20030519001	
TIO-540-J2BD TO-540-J2BD	8520	BRACKET CRANKSHAFT	13F17786	BROKEN BROKEN	20030616001 20030529001	QUE
10-540-J2BD	8520	CRANKSHAFT	LW12163	CRACKED	20030529001	
SOMBARDIER RO		CIVAINIONAFI	LVV 12 103	CIVACILL	20030000007	AIL
RECIPROCATING		EXHAUST VALVE		DESTROYED	20030605002	PNP
912 S	8530	INTAKE VALVE		SEPARATED	20030003002	
J J	3000	41 7/16/6		J / II V II LD	_55555521504	2111

MAKE/MODEL	ATA	PART NAME	PART NO. PA	ART	CONDITION	SDR NO.	RGN
CURTISS WRIGHT R-1820-80	Г 8530	CYLINDER	893742C		PERFORATED	20030605007	ONT
GARRETT TFE731-3 TPE331-10N TPE331-10N TPE331-10UA TPE331-10UA TPE331-10UGR516H TPE331-6-252B		POWERPLANT O-SPRING TOGGLE SWITCH BUSHING FUEL CONTROL UNIT TTL BYPASS VALVE TORQUE SENSOR	30728001 MS3505929 310170318 UNI8978017 8974573 31017253		BIRD INGESTION BROKEN INTERMITTENT NEW INTERMITENT UNKOWN	20030528002 20030605005 20030526010 20030512002 3 SDRs 20030618008 20030414003	PNR PAC PNR PNR PNR
ORENDA OE600A OE600A	7420 7920	IGNITION HARNESS HOSE STRAP	T20016 7033012		HEAT DAMAGE BROKEN	20030404006 20030404007	
PRATT & WHITNE JFTD12A-4A JT15D-4 JT8D-9A JT8D-9A PT6 PT6A-135A PT6A-28 PT6A-28 PT6A-28 PT6A-22 PW120 PW120A R-1830-92 R-2800-79 R-2800-79 R-2800-79 R-2800-79 R-985	7230 7320 0000 7930 7321 7261 7261 7310 7921 8530 0000 8520 8530 8530 8550	FUEL PRESS & DUM GOVRNOR LEVER ELECTRIC RECEPTA BY-PASS SWITCH MAIN BODY O-RING ENGINE BY-PASS VALVE WIRING ENGINE OIL COOLER FITTING OIL COOLER CYLINDER CYLINDER MAIN BEARING ROCKER JAM NUT SPARK PLUG NILUNIKNOWN	586982 AS3240070 BACC45FN12C33 1163P181 2525098 AS3209017 SE3024566 87900014103 28E99X 114651 98181	S	FAILED SEPARATED SHORTED FAILED MACHINED DOWN SPLIT FAILED WORN FAILED BROKEN LEAKING CRACKED FAILED FAILED LOOSE LOOSE	20030609005 20030628008 20030627008 20030627008 20030527002 2003051007 2 SDRs 20030613008 20030613008 20030630005 200306221007 20030627006 20030627006 20030423001 20030527006 20030527006 20030527006 20030527006	QUE ONT QUE QUE PAC ATL PNR PNR ATL QUE VAR ONT PNR PNR PNR PNR PNR PNR PNR
R-985-AN-14B R-985-AN-14B 2800CA3	8520 8550 8550	REAR CASE RECIPROCATING EN NOSE CASE	126299 519589		CRACKED INSERT MISSING	20030605018 20030609006 20030604005	PAC PAC
ROLLS ROYCE BR700-715A1-30 RE211TREN1772B60 RB211-22B-02 RB211-534E4 SPEY 511-8 SPEY 511-8	7532 7230 7261 7120 7250	HP COVER PLATE IP VALVE ENGINE HP/IP TURB. BEARING RING LOCKING TRU BOLT	BRH12388 HTE4500019 UL39099 PEP6295 EU24036		DISINTEGRATED HINGE BOLT BROKEN FRACTURED CRACKED BROKEN	20030501006 20030522001 20030526002 20030619001 20030611004 20030507005	QUE QUE QUE NCR
TELEDYNE CONT IO-360-ES IO-470-C IO-470-L IO-470-L IO-520-D IO-520-MB O-200-A O-200-A TSIO-520-AF TSIO-520-E UNKNOWN	2434 8530 8530 8550 0000 0000 0000 8011 8520 8560 8011	MOUNT BRACKET CYLINDER PISTON ASSEMBLY VALVE LIFTERS ENGINE MOUNT CON ROD SET SCREW STARTER CRANKCASE ASSEMBLY TURBOCHARGER STATER SHAFT GEAR	654358 625859CP AEC648029 646277 646476 21153 Z4214 643989 4066109025 AEC639389		CRACKED CRACKED BROKEN NEW CRACKED BENT LOOSE BLOCKED WORN CRACKED FAILED	20030513001 20030430004 20030423005 20030506003 20030627003 20030627007 20030514012 20030409005 20030609007 20030613002	PNR PNR PNR PAC PAC ONT QUE PAC PNR
TURBOMECA ARRIEL 1B ARRIEL 1D1 ARRIEL 1D1 ARRIEL 1D1	7230 7250 7421 7421	ENGINE 2ND STAGE TURBINE IGNITER IGNITER	0292253350 9550168760 9550175400		RUBBING CREEPING U/S U/S	20030514005 20030424003 20030528014 20030610008	PAC PAC
propelle	: -						
DOWTY ROTOL R333/4-82-F/12 HAMILTON STAND 2D30-301	6114 DARD 6122	PISTON SEAL O-RING PROPELLER GOVERN1/	A2G5		BROKEN FAILED	20030407002	:
HARTZELL HC-B3R30-4B HC-B3R30-4B HC-E2YR-2B HC-E2YR-2B HC-E3YR-2ATF HC-E4A-3D	6100 6100 6111 6114 6114 6140	NUT NUT BLADE PROP ATTACH BOLT BEARING WIRE HARNESS	B1894 B1894 FC84754 SA2202 30600003		CRACKED CRACKED CORRODED FAILED WORN SHORTED	20030605016 20030605017 20030604006 20030606001 20030604008 20030514009	PAC PAC PAC PNR

MAKE/MODEL	ATA	PART NAME	PART NO.	PART CONDITION	CTRL NO. RGN
MCCAULEY 2AF34C55 SENSENICH W96JA-56	6114 6114	HUB/NUT SAME	D4883 SAME	CRACKED CRACKED	20030530006 ONT 20030527001 ONT
equip <i>m</i>	ent				

AMERIKING						
AK450 ARTEX	0000	BATTERIES		CORRODED	20030627006	ONT
ELT1104	2562	BATTERY PACK	4520130	CRACKED	20030605011	ONT
AUOFLUG URHB37E AVIALL	7410	SPARK PLUG(AUTOL	URHB37E	BROKE IN HALF	20030411003	ATL
MS24665143	1430	COTTER PIN	MS24665143	DEFORMED	20030530007	NCR
B F GOODRICH 211333 BEECH	0000	BRAKE HOUSING	2158750218	CRACKED	20030619008	NCR
3536415157 BENDIX	2752	FLAP MOTOR AND	D3536415157	UNSERVICEABLE	20030506006	PNR
430121200 430121204 GE	7323 7323	PT GOVERNOR PT GOVERNOR			20030609013 20030609014	
2CM70D6A	2434	GENERATOR	2CM70D6A	OVERHAULED	20030506004	PNR
GOODRICH 101161 ISRAEL	3246	INNER WHEEL HALF	101161	CRACKED	20030603004	PNR
25W78301601	2120	RAM AIR DUCT	25W783016011	U/S	20030424004	NCR
KOLLSMAN 31400117	0000	STANDBY ALTIMETER			20030625005	NCR
230790021 23079009	2435 2435	BEARING STARTER/GENERATOR	03601018	DISINTEGRATED	20030421010 20030407003	
MARVELSCHEB 105193 MESSIER	7322	IDLE MIXTURE		SCRMISSING	20030421002	ONT
82970014011 82970018009	3232 3232	OUTER CYLINDER CYLINDER ASSY	82970015005 82970018009	CRACKED WORN & CORRODED	20030516004 20030516003	
PILATUS 5005012117	2432	BATTERY HEATING	TAS203	BURNED	20030422003	ONT
POINTER PS400010	2562	EMERGENCY LOCATOR			4 SDRs	PNR
PRESTOLITE MZ4218	8011	ARMATURE	MZ2399S	BROKEN	20030506002	ONT
SLICK 4371 4372	7414 7414	MAGNETO IMPULSE COUPLING	M3100	LOOSE RIVET	20030425001 2 SDRs	ONT ONT
UNKNOWN MARKSGONE	0000	RELAY	MS24171D1	SHORTED	20030628001	PNR

<u>LEGEND</u>

ATA Air Transport Association number defining assembly/system/component
SDR NO. TCA assigned SDR control number - please quote in any correspondence or inquiries
TCA region of SDR submitter:

PAC = Pacific, PNR = Prairie Northern,
ONT = Ontario, QUE = Quebec, ATL = Atlantic,

NCR - Ottawa (HQ), VAR - more than one Region

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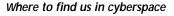
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