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Feed*Back* is published quarterly by the Continuing Airworthiness Division of Transport Canada, informing the aviation community of reported day-to-day problems that affect aircraft airworthiness in Canada.

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The articles contained in **FeedBack** are derived from Service Difficulty Reports (SDRs) submitted by Aircraft Maintenance Engineers (AMEs), owners, operators and other sources in accordance with *Civil Aviation Regulation* (CAR) 591.

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All defects or occurrences should be reported to Transport Canada through the Service Difficulty Reporting Program. For additional information about this program or concerning an article in feedback magazine, contact your nearest Transport Canada Centre.

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TP 6980E

FIXED WING

BRITISH AEROSPACE BAE 125-800A

SDR # 20061104001

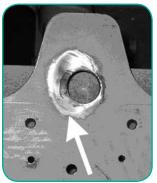
Reinforcement Plate Corroded

Upon visual inspection and cleaning of attachment lug area of a vertical fin to horizontal stabilizer, the technician noticed material flaking off of one bolthole area. Further inspection revealed exfoliation corrosion. Since this type of corrosion is difficult to detect visually it was only when the technician was cleaning the area that he noticed the damage.

Upon consultation with the manufacturer (Raytheon), the corrosion was removed to determine the extent of penetration and coverage (photo illustration). The penetration and coverage was determined to be out-of-tolerance and required replacement of the reinforcement plate. This was the first time the manufacturer had come across this type of discrepancy.

Especially significant was the fact that it was exfoliation corrosion and not any type of surface corrosion. The task to replace the part was extensive, requiring specialized jigs, tools and machining along with rigging checks. The job was successfully completed by a qualified service center.

One of the most destructive forms of corrosion is exfoliation corrosion. Exfoliation corrosion follows grain boundaries. It occurs in multiple planes, causing a leaf-like separation of the metal grain structure. This form of corrosion causes a loss of load-carrying capability. The most effective way to control this kind of corrosion is to have a grain structure that is not susceptible to exfoliation.



Transport Canada recommends that operators and maintenance organization be cognizant of the manufacturer's Corrosion Prevention and Control Program (CPCP) and aging aircraft initiatives.

Corroded (exfoliation) area blended out – Out-of-Tolerance

DASSAULT FALCON 50

SDR # 20060615002

Horizontal Stabilizer - Guide Plate Disconnected

The horizontal guide plate located in the area of the horizontal and vertical stabilizer interface was found detached. Closer observation revealed that the guide plate was bent and lying freely on top of a rib integral to the horizontal stabilizer assembly.

Fortunately, this situation was discovered because there was a potential of the detached guide plate interfering with

the elevator control linkage. This defect was noted when a technician required access to this area for another reason.

The guide plate was repaired in accordance with the manufacturer's instructions and then reattached to the horizontal stabilizer fillet assembly.

Maintenance personnel with extensive experience on Falcon aircraft had not seen this situation before. The submitter recommended that the guide plate rivets be closely inspected whenever the spring fillet assembly is removed.

The spring-loaded horizontal stabilizer fairings are located at the horizontal stabilizer and vertical stabilizer interface and move with horizontal stabilizer trim inputs, providing aerodynamic functionality. The spring loaded fairing assembly can be difficult to install and requires close attention to detail because of attached hardware. The guide plate is attached to the fairing but is not visible unless the spring-loaded fairing is removed for inspection or other maintenance activity. Close attention to this area is recommended.

DASSAULT FALCON 2000

SDR # 20060911006

Incorrect Light Bulb - Missed It by One Number

While performing the post flight inspection, an electrical burning odor was detected in the aft area of the cabin. Further inspection of the lavatory mirror vanity lights was carried out and one lamp was found to be defective. When the lamp was removed for replacement, it was noticed that one end of the lamp assembly had started to melt. With the defective lamp assembly removed it was also noted that the defective part number was incorrect. The lamp found installed was AL-845-T-279 however, lamp assembly AL-845-T-279-F should have been installed as per the Falcon 2000 IPC.

After the defective lamp assembly was removed and tested, the electrical burning smell in the cabin was no longer present.

Extreme care must be used when installing interior or exterior lamps. The smallest deviation in part number could heat things up. **

DE HAVILLAND DHC 2

SDR # 20061016004

Float Strut Stress Cracks

During 100-hour inspection of a DHC 2 (Beaver), six (6) inch stress cracks were found to have a originated at the top forward side of the struts on the right and left rear float struts.

Time-in-service is unknown and total aircraft time is over 22,000 hours. The aircraft is operated exclusively on floats in a corrosive environment and generally adverse landing conditions.

Most of these legacy aircraft have seen many hours of operation in a corrosive environment and rough water landings are common. The deterioration of aircraft structural elements can be greatly accelerated by the environment and undesirable landing conditions. Special attention to these legacy aircraft will keep them structurally sound and safely flying for many more years.



DE HAVILLAND DHC 6

SDR # 20060920010

Float Struts Broken

A foreign operator of a DHC-6 Twin Otter reported to the aircraft manufacturer that he landed in high waves and broke both front float struts. The aircraft taxied to the dock without incident.

A maintenance technician inspected the aircraft for other damage caused by landing in such rough water. No other damage was evident. Both front float struts were replaced and the aircraft returned to service.



The aircraft manufacturer has had reported incidents derived from floatplanes landing in extremely rough waters. Recently the manufacturer has issued service bulletins and will revise the Maintenance Manual due to reported engine mount failures. These failures were a consequence of extended operations in rough waters.

Transport Canada recommends following the manufacturers' recommended operational procedures and inspection criteria.

When operations require you to utilize the aircraft in extreme conditions, preventive maintenance and additional inspections may have to be considered. *

DE HAVILLAND DHC 8

SDR # 20061026002

Aileron Control Cable Worn

During the C-check inspection, five out of eight cable pulley bearings within the wing aileron system were found seized and had flat spots

All cables passing through these pulleys had to be replaced due to the chafing wear; the worst one was cable, part number 82700519-001.

Depending on your operation and environmental conditions in which you operate, these pulley bearings may require a more frequent inspection.

- Inspect the pulley bearing to ensure it is not seized and the pulley rotates smoothly.
- Inspect the condition of the pulley groove to confirm that it is not excessively worn and has no flat spots, which could cause the pulley to jam and not rotate.

The cable part number 82700519-001 is Item 440, Figure 25, on page 84 of the Dash 8-100 Aircraft Illustrated Parts Catalogue PSM 1-8-4. A picture of this badly worn cable and pulley is illustrated.





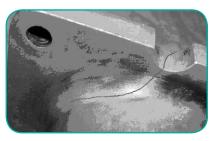
DE HAVILLAND DHC 8-102

SDR # 20061103002

Mounting Adapter Cracked

During a C-check inspection, a crack was discovered on

the rudder actuatormounting adaptor (upper) to the vertical stab spar. The crack is approximately 2.5 inches around the circumference of the adaptor tube.



Although there is no probable cause mentioned, it is likely stress induced fatigue cracking. A thorough visual inspection



by the technician detected this crack and prevented subsequent failure of the mount.

Passenger Call Button Circuit Board Shorted

Upon arrival at the main base for overnight maintenance, flight attendants alerted maintenance personnel that a number of the passenger call button lights could not be reset. During the course of the maintenance check, it was discovered that all of the passenger service unit (PSU) control circuit boards had heat damage.

Investigation into the cause of the circuit board damage revealed that the wire on the outlet side of all of the PSU control circuit boards had chafed through the outer insulation resulting in a short circuit to ground at STN X312.35. This short circuit resulted in the outlet transistor to burn out on all PSU control circuit boards.

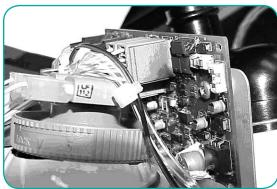
All aircraft cabin PSU's were replaced and the aircraft returned to operational service.

It is to be noted that since the aircraft was brought into service, the area of the wire chafe had not been inspected nor was it a requirement. The damaged wire was able to contact a trim support clip because the wiring bundle was incorrectly placed outside of the cable trough.

The aircraft manufacturer has indicated that an investigation into this incident was conducted. It has been determined that the wiring bundle should have been in the wire trough provided. Also, as per the drawings, no insulation clip is located in this area.

The manufacturer is considering the addition of a fuse to prevent damage to the PSU if an electrical short in the wiring installation were to occur.





STN X312.35 - Wire 3322AP - Cable trough - Insulation retainer Clip

Copilot's Display Unit Short-Circuited

During flight, the crew reported that the copilot's primary flight display (PFD) went blank, then came back on in post mode (green T) and then went permanently blank again.

Following this incident a burning odor with no visible smoke or flames was detected. The odor was carried throughout the cabin by the re-circulation fan and detected by the flight attendants. Mechanics found evidence of soot around the cooling holes of the removed PFD.

The part was returned to the PFD manufacturer for investigation. The manufacturer reported a strong burnt smell before disassembly. When the power was turned on to the PFD unit there was high smoke emission. The root cause analysis was the short circuit on C43. This type of ceramic capacitor (C43) is no longer used on power supply module (PSM) boards.



Thales Avionics (manufacturer) issued a service bulletin (SB) C19190A-31-014 to inform operators of the introduction of a new technology of capacitors in the PSM board in order to prevent capacitors from breaking and causing short circuits.





Rudder Attachment Assembly Damaged

While performing an aircraft walk around, there was excessive play found at the rudder upper attachment bolt.

Further investigation discovered the upper attachment bolt had elongated the hole and also damaged the nose rib. When trying to remove the bolt, the nut plate broke away and a piece of the nose rib that was already cracked actually broke away with the nut plate. The bolt head had to be cut off to remove the rudder and the upper bearing was found damaged. The damaged rudder is now awaiting parts to facilitate a repair.



Maintainers are reminded that movement of flight controls while performing daily checks can detect latent failures. Had this defect gone undetected, serious rudder control restrictions could have been experienced. 🛠

Propellers

HAMILTON STANDARD PROPELLER 14SF-7 (DHC 8-102) SDR # 20060905006

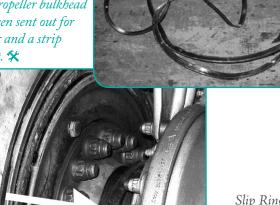
Brush Block Slip Ring – Bulkhead Separation

The flight crew snagged the LH propeller de-ice system as unserviceable as there was no indication on the electrical load gauge.

Upon further investigation, maintenance personnel found that all three of the propeller brush block slip rings had separated from the propeller bulkhead and then wrapped around the propeller shaft.

The surrounding area of the engine compartment was inspected but no damage was found.

The propeller bulkhead has been sent out for repair and a strip report. 🛠



Slip Rings

HARTZELL PROPELLERS HCB5MP

SDR # 20060926005

Propeller Blade Retention Shanks - Corroded

A visual inspection revealed corrosion on the blade shanks of all the five blades that were removed from Shorts

& Harland SD3-60 aircraft. Even after rework to the specified minimum dimensions, the corrosion damage was still present. It was noted that the blade width and thickness was well above minimum dimensions.

All five propeller blades were removed from service.

Environmental factors may have contributed to the excessive corrosion found on the blade retention shanks. 🛠



WOODWARD PROPELLERS

SDR # 20060804006

Propeller Governor - Idler Gear Bushing Worn

The engine was submitted for its scheduled first time overhaul. When the propeller governor 1584 hours time since new (TSN) was dismantled for cleaning before overall inspection, it was discovered that the idler gear bushing, P/N 5337050, was severely worn. The excessive bushing wear resulted in an elliptical rotation of the idler gear, which caused the gear pocket to wear in an oval manner on the pressure side.

A normal idler gear bushing measures approximately 0.100",

however this worn idler gear bushing measured 0.0175". It is not known what caused this excessive bushing wear.

The periodic oil analysis did not detect any metal contamination. After discovering this condition, the maintainers were unable to rotate the



gears manually and found that the gears would stick and bind. This raised concerns that if this occurred in service, then seizure and complete loss of propeller governor oil pressure (loss of thrust) was possible. The governor manufacturer stated that they seen a similar wear condition before on a TPE 331 engine which had a propeller governor of similar design.

ENGINES

GENERAL ELECTRIC CF34-3B1 [CL600-2B19 (RJ200)] SDR # 20060728004

Fan Blade Liberation

While climbing out to 18,000 feet, the cockpit crew heard a loud bang, followed by high vibrations on the #1 engine. The fire warning initiated and the pilot discharged the fire bottles and shut down the engine. An emergency was declared and the aircraft returned to departure airfield and performed an uneventful single-engine landing.

Maintenance personnel reported that one of the fan blades was missing and that the jet pipe appeared burnt and cracked.

The General Electric field service representative observed that one fan blade and the outboard fan cowling was missing. Two (2) small punctures were noted in the fan containment case as well as damage to the bolts that connect the LP Turbine and HP Turbine flanges. Fire damage to the exhaust nozzle was evident.

Shortly after this event, the engine manufacturer issued an All Operator's Wire CF34-06-07 related to this event. Investigation is ongoing at this time to determine the cause of failure of the #11 fan blade.



HONEYWELL (ALLISON) 250-C20B

SDR # 20060927002

Fuel Control Unit Fuel Filter - Contaminated

The engine-driven fuel pump filter was replaced with a new PMA filter which was pre-packaged in plastic wrap. This particular PMA filter has an opening at one end and is sealed at the other end. The installer used his finger to remove the plastic wrapping, however a piece of plastic inadvertently got pushed up inside the fuel filter.

During a functional check when fuel was introduced into the system, the plastic remnants traveled up the fuel line and lodged inside the fuel control unit (FCU) fuel filter.

The SDR submitter stated that the Purolator airframe fuel filters are also wrapped in this fashion. The main difference is that the airframe fuel filter is open at both ends allowing pieces of plastic to fall though it. Maintainers must be aware of this problem and ensure that filters are free of contaminates prior to installation.

Transport Canada Civil Aviation concurs with the SDR submitter's recommendation that maintainers ensure that filters are clear of contaminates. Always double-check before installing parts. **



HONEYWELL GTCP36-150RJ (CL600 2B19)

SDR # 20060604002

Honeywell APU-Smoke/Flames

While taxiing out to the runway for takeoff, the crew noted a slight exhaust/fuel odor. Shortly thereafter, the crew was advised that a flame was seen coming out of the APU exhaust. The crew did not have any indication of an APU fire. However, the crew shutdown the APU and taxied off the active runway. Shortly after stopping the aircraft, an APU Fire Fail and APU Fire Warning message was received on EICAS (Engine Indication Crew Alert System). The crew discharged one fire bottle into the APU area.

The airfield firefighters arrived and discharged halon into the APU enclosure while the passengers were safely evacuated. The APU was replaced and the aircraft returned to service.

The APU teardown investigation revealed there was a fuel leak/separation at the #2 atomizer fitting caused by a high cycle fatigue (HCF) crack in the tube. Additionally, the lower B-nut on the fuel solenoid was leaking. The HCF crack initiated from areas on the outside diameter of the fuel tube and was produced as a result of reverse bending.

Heads un

It appears that the bending of the fuel tube was a contributing factor in the APU fuel leakage and resultant fire. Proper maintenance of fuel lines and fittings is critical because of the flammability of fuel. Even a small leak in a confined area can produce an explosive atmosphere, which can be ignited by any kind of spark.

PRATT & WHITNEY CANADA PT6A-67B [CL600 2B19 (RJ100)] SDR # 20060918009

Engine Thrust Reverser – Track Cracked

While carrying out an inspection on the engine thrust reverser system, the AME visually detected a significant crack on the LH lower thrust reverser track.

Reference: CSP A-006 IPC Chapter 78-34-11, Figure 1, Item 75A & 95A.

The cause of the thrust reverser crack is likely due to time in service.



HEADS UP

Exhaust Type Heaters & Carbon Monoxide Hazards

One of the most common types of aircraft interior heaters is the exhaust type shroud (muff) heater that is primarily used in reciprocating, single engine aircraft.

This is a simple design that uses exhaust heat transfer. A shroud (muff) is placed around the exhaust stack and during flight the outside atmospheric ram air is forced through the shroud (muff) and around the exhaust stack thus allowing heat to transfer ducting and then into the aircraft interior. While in flight, the pilot can regulate the interior temperature by using an alternate source of atmospheric ram air.

One of the inherent dangers of the exhaust type heater is the possibility of carbon monoxide poisoning (CO). CO gases are colorless, tasteless and odorless, thus becoming even more insidious as they become mixed with other engine fumes/gases, while operating in the aircraft environment. Long exposure to low CO concentrations is as hazardous as short exposure to high concentrations.

Personnel are reminded to remain current with the inspection requirements detailed in Transport Canada Airworthiness Directive CF-90-03R2 titled "Exhaust Type Cabin & Cockpit Heaters". Additionally, it may be prudent to conduct more frequent pre or post flight

inspections, paying particular attention to exhaust stack welds/seams and for possible exhaust leaks inside the shroud (muff). Some aircraft manufacturer recommends that exhaust manifold and heater assembly be inspected as frequently as every 25 hours. CO can also seep into the aircraft interior through openings in the firewall or cabin fuselage or heating system. The danger of CO poisoning during winter operations is heightened because the fresh air vents and windows are normally closed.

Currently, there are two types of CO detectors available to measure CO concentrations in the aircraft capsule. One type of detector draws a sample of air into a tube containing material that will change color according to the amount of CO that is present. Another type of CO detector uses a porous plastic disc that contains a chemical, which will change colors according to the concentration levels of detectable CO.

Transport Canada recommends that operators install CO detectors in their aircraft.

For further information, it is highly recommended that operators also refer to *Federal Aviation Administration* (FAA) Advisory Circular (AC) 20-32B located at http://www.faa.gov. **

EQUIPMENT ADS

Transport Canada (TC) endeavours to send copies of new airworthiness directives (ADs), which are applicable in Canada to the registered owners of the affected products. Equipment/appliance ADs are often only distributed to our regional offices because the owners of aircraft affected by this type of AD are not generally known.

The following new ADs on equipment have been received by TC in the last three months. AMEs and operators of the affected products are encouraged to obtain further information or a copy of the ADs from their regional TC office, their local TCC, their PMI, or from the Civil Aviation AD website at: http://www.tc.gc.ca/aviation/applications/cawis-swimn

AVIO INTERIORS	2006-0264	EU	Equipment & Furnishings - Passenger Seats Rear Fitting
CALEDONIAN AIRBORNE	2006-0241	EU	Replacement of Main Case P/N BC85-051 and piston BC85-052 with improved units made of stainless steel 6S80D
GOODYEAR	2006-18-08	US	Goodyear Aviation Tires - tread separations and tread-area bulges
HONEYWELL	2006-19-04	US	To prevent the transponder of the COM unit from going into standby mode
INTERTECHNIQUE- ZODIA	2006-0286-E	EU	Oxygen Reserve Cylinders - Removal/Emptying
PARACHUTE SHOP	2006-0279	EU	Equipment/Furnishings - Parachute Equipment - Removal from Service
RECARO A/C SEATING	2006-0220	EU	Equipment - Passenger Seats - Inspection of Seatbelt Shackle
SANDEL AVIONICS	2006-16-18	US	Bearing error caused by input fault & software error - Install Placard, Revise Flight Manual and update Software.
STC ST02129AK	2006-15-12	US	Carrying both cargo and passengers in the same compartment
STC ST02177AK	2006-15-11	US	Carrying both cargo and passengers in the same compartment
TECNAM	2006-0234	EU	Equipment & Furnishings - Seat Rail Stops – Inspection

FAA SPECIAL AIRWORTHINESS BULLETINS (SAIBS)

An SAIB is an information tool that alerts, educates, and makes recommendations to the general aviation community. It is non-regulatory information and guidance that does not meet the criteria for an Airworthiness Directive (AD). http://www.faa.gov/aircraft/safety/alerts/SAIB/

SAIB#	Manufacturer	Model	Issue Date DD/MM/YY
SW-07-10	Eurocopter France	EC 155B and EC155B1 helicopters	11/06/2006
NE-07-09	Teledyne Continental Motors (TCM)	IO-520, TSIO-520, IO-550, IOF-550 engines	10/31/2006
CE-07-08	Cirrus Design Corporation	SR20 & SR22 airplanes	10/31/2006
	Columbia Aircraft Manufacturing	LC41-550FG & LC42-550FG airplanes	
	Piper Aircraft Inc.	PA-28, PA-32, PA-34, PA-44, PA-46 airplanes	
CE-07-07	Socata	TBM 700 airplanes	10/27/2006
CE-07-06	General Aviation Aircraft	Alcohol (ethanol or methanol) present in the automobile gasoline	10/27/2006
SW-07-05	FH-1100 Manufacturing Corporation (Siam Hiller Holdings, Inc.)	1100 (OH-5A) and FH-1100 helicopters	10/27/2006
CE-07-04	Grob	G120A airplanes	10/18/2006
NM-06-54R1	Transport Category Airplanes	Handheld fire extinguishers	10/18/2006
NE-07-03	Pratt & Whitney Canada (P&WC)	JT15D-5 series engines	10/12/2006
CE-07-02	Garmin	WAAS receiver equipment; GNS 480 and CNX80 Navigation System	10/12/2006
NE-07-01	General Electric Aircraft Engines (GE)	CF6-80C2 and CF6-80E1 series turbofan engines	10/12/2006

SERVICE DIFFICULTY REPORTS

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NCR = Ottawa (HQ) ATL = Atlantic ONT = Ontario QUE = Quebec VAR = More than one Region TCA assigned SDR control number - please quote in any correspondence or inquiries Joint Aircraft System Code number defining assembly/system/component TCA region of SDR submitter:

PAC = Pacific PNR = Prairie and Northern SDR NO. JASC RGN

Make/Model	Jasc	Jasc Part Name	Part No.	Part Condition	SDR No.	RGN
AIRCRAFT						
AERO COMMANDER 690	5540	FITTING	4200821	WORN/LOOSE RIVET	20060714004	PNR
AEROSPATIALE	:					
AS 350B2 AS 350B2 AS 350B2	5210 6220 6720	DOOR SEAL YOKE ASSEMBLY STOP FRICTION CHINE SHIM (PAD)	852G10 350A37116200 350A75111720	DEPARTING FAILED SERVICARIE	20060803001 20060922002 20060710005	PAC PNR PAC
AS 350BA AS 350BA	2913 6310	SLEEVE COUPLING FREEWHEEL ASSEMBLY	S40 35013001	UNSERVICEABLE MAKING METAL	20060804007 2 SDRs	PNR PAC
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AIR TRACTOR						
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AIRBUS						
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A340 313 A340 541	3444 2500	TERRAIN SWITCH	9650976003206	BURNT WIRES BURNT	20060725011 20060801007	OUE OUE
BAE - UK						
3112	3246	INNER WHEEL HALF ASSY	300720	SERVICEABLE	20060719005	PNR
BAE 146 200	2740	LEVER ARM	HC273H0344	FAILED	20060810007	NCR
BAE - USA						
HS 125 700A	5280	DOOR ASSY - RH	5UDZ198	CRACKED	20060906011	NCR
BEECH						
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PART NAME		STARTER GENERATOR FITTING-CROSS FITTING FILLER CAP (LANYARD) F506 FILLER ASSY INBOARD SKIN HOSE PENDULUM WEIGHT WASHER TAILBOOM SHEAR BEARING SUPPORT SPINDLE ASSY CONNECTOR 1B14P1 CHECK VALVE	BOLT MAIN ROTOR BLADE ACORN NUT HANGER BEARING TYR HYD. SERVO ACTUATOR DRAIN VALVE PRESSURE SWITCH	CONNECTOR O-RING MUICRO SWITCH START VALVE INDICATION SWITCH TRANSFORMERE RECTIFIE FIRE DETECTION CONTROL BEARING FRAME PULLEY HYDRAULIC LINE LOW PRESSURE SWITCH OFF-WING ESCAPE SLIDE
TASC	3210 5610 7712 2750 5544 000 3230 7922 3236 5311	2435 2435 2910 5302 6320 0000 2910 6220 1000 5302 6220 6220 6220 6220	6720 6210 6220 6510 6730 7170 7931	3241 2740 5230 7312 2781 2730 2710 2710 2710 2710 2710 2710 2710 271
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RGN QUE QUE QUE QUE ATIL ATIL NCR	OUE PNR PNR PAC
SDR NO. 20060825001 20060818002 20060819004 2 SDRs 20060919004 20060918009 20060925005 2 SDRs 8 SDRs 8 SDRs 2006072001 20060704020 20060704020 20060713003 20060713003 20060830002 4SDRs	20060828001 20060712003 20060712003 20060825003 20060810004 20060910001 20060913001
PART CONDITION CHAFED U/S U/S NEW CRACKED CRACKED U/S CRACKED FAILED FAILED FAILED CRACKED FAILED CRACKED FAILED CRACKED FAILED CRACKED FAILED CRACKED DETACHED LEAKING	BROKEN SHAFT BURNT AND MELTED CORRODED CRACKED U/S CRACKED CRACKED CRACKED CRACKED CRACKED CRACKED CRACKED BROKEN BEPARATION CRACKED DISCOLORED DISCOLORED DISTORTED ON CONDITION CRACKED EAILED WORN OVER HALF SEPARATED DELAMINATED CORROSION SHORTED FAILED FAILED
PART No. 285T003117 BR90002 60021134434 NP1393225 CF343A1 22850899119 GG670950099 NP15932211 4120T16P01 16103000 AN62498 66WA300 327628	6052T06P05 2XA80ABBLU 04310093 053200198 D4FF10316GA RA105002 05533018 SLC36005F 07503214 SLC36005F 0712024 05523311 07221991 26150181 26150181 26150182 26150182 26150182 26150182 26150183 26101133 230760011 7020362 656530129 95442077 991214736 52771338 S35979
PART NAME WIRE W440-001-18 WARNING CARD EICAS MESSAGE BALLAST VISOR ASSY L/H WINDOW ENGINE TRACK SCAVENGE SCREENS AIR CYCLE MACHINE L/H SIDE WINDOW SCREEN OIL PRESSURE TRANSDUCER INNER WHEEL HALF HYD CHECKVALVE PUMP FACE ATTACH CYLINDER NOZZLE EXHAUST FUEL CELLS	ENGINE FUEL PUMP GENERATOR FEED CABLE WIRE GENERATOR FEED CABLE WIRE 22QA WIRE BRAKE LINING RIVET L/H DRIP ASSEMBLY OVERVOLTAGE CONNECTION CYLINDER BULKHEAD CYLINDER ASSY SEAT BACK FRAME MAIN GEAR LEG BUSHING FWD PROP SPINNER BULKHEAD RIB TRIM PANEL DEICE BOOT FAIRING WHEEL HALF - OUTER FAIRING WHEEL NOSE WHEEL
JASC 2913 2200 752 3300 7520 7200 7200 7231 2100 5610 2330 7931 2246 22913 8530 7800 2810	5530 5530 5531 2497 3246 5710 0000 0000 0000 3221 6113 5712 2710 3246 2712 2720 3246 2713 273 2746 273 3246 2713 2714 2714 2714
MAKE/MODEL 767 333 767 375 BOMBARDIER CL600 2B19 (RJ100) CL600 2D19 (RJ700) CL600 2D19 (RJ700) CL600 2D14 (RJ700) CL600 2D15 (RJ900) BRITTEN NORMAN BN2A 26 CANADAIR CL215 1A10 CL215 1A10 CL215 1A10 CL215 1A10 CL215 GB11 (CL215T) CL215 GB11 (CL215T)	CL600 2A12(601) CL600 2B16(601 3A) CL600 2B16(601 3A) 150M 172K 172M 172M 172P 172P 172P 172P 172P 172P 172P 172S 185D 208B 208B 208B 208B 208B 208B 208B 208B

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RGN PAC ONT ONT PAC ONT PNR ONT	PAC PAC PAC	PNR PAC
SDR No. 20060817005 20060817003 20060825004 20060906026 20060705005 20060825002 20060717003 20060808006 2006081001	20060928003 20060706007 20060911002 20060817001	20060911006 20060822008 20060724002 20060920007 2 SDRs 20060801009 20060801009 20060801005 20060818001 20060818001 20060818001 20060818001 20060816002 20060816002 20060816002 20060918011 20060918011 20060918011 20060918011 20060918001 20060918001 20060918011 20060918001 20060918001 20060918001 20060918001 20060918001 20060918001
PART CONDITION BROKEN LEAKING GROUND DOWN SHATTERED U/S SEVERAL CRACKS CRACKED CRACKED CRACKED UNSERVICEABLE CRACKED	FAILED GUSSET CHAFED DETERIORATION SEVERLY WORN	MELTED CORORRED/LEAKING FAILED BROKEN U/S U/S U/S U/S U/S U/S CRACKED BURNT RUPTURED CRACKED CRACKED CRACKED OWN WITHOUT CSP CHAIN FRACTURED U/S
PART No. 01073370 25630005 183238001 991438012 S18802 075100331 12116011 S122210 075302781	222182 89H1014738LGD4 AN4H12A	AL845T279 C2P2009 C5P156171 C6UF10151 BAO80061 734187B A44700009 8800121 2431154B10 82950010141 2890410115 DSC252A40230 82700562005 AN81510D BV33001215 S2L354 85220270001 MS2764639 83231044005 82970009325 L312111 DSC5108 6617303 478441 216052 82742409-001 AND 410-001 304850901
PART NAME INERTIA REEL SHAFT SEAL TR SOLENOID WINDSHIELD LIGHT RHEOSTAT HINGE ASSEMBLY-STABILIZER ENGINE MOUNT MLG OUTBOARD FITTING THROTTLE CONTROL TORQUE PLATE	STRINGER SUPPORT LDG GEAR DOWN LINE EXPLOSION SUPPRESANT FOAM ATTACHMENT BOLT	LAMP ASSEMBLY PIPE FLANGE RING GENERATOR FRONT FLOAT STRUT BALLAST TORQUE TUBE SERVO ACTUATOR NLG SHOCK STRUT WIRE HYDRAULIC LINE HYD LINE ASSY FLEXIBLE HOSE ASSEMBLY CABLE CHAINE ASSY HYD UNION FITTING LAMP HOLDER WEATHER RADAR INDICATOR O-RING MECHANISM EMERG EXIT ROD ENDS GUARD ASSEMBLY LINE OUTER BEARING CUP BEARING ENGINE DRIVEN HYD PUMP NLG WOW 2/ CENTERING HARNESS NLG DOOR SPRING BRAKE UNIT
JASC 2510 7830 7322 5610 3300 5520 7120 5280 7322 3242	2910 0000 7300	3300 2820 7800 2434 3246 3300 2700
MAKE/MODEL 560 560 750 750 A185E A185F U206B U206B	CHAMPION TECA CONAIR FIRECAT TURBO FIRECAT CONVAIR - CANADA 340	DASSAULT FALCON 2000 DE HAVILLAND - CANADA DHC 2 MKI DHC 2 MKI DHC 6 DHC 8 100 DHC 8 100 DHC 8 102 DHC 8 102 DHC 8 102 DHC 8 102 DHC 8 300 DHC 8 301 DHC 8 400

RGN	ATL	PNR	ONT ONT OUE	ONT PNR ONT	PNR	NCR	ATL ATL	PAC PNR PNR PAC NCR	PNR	ONT ONT ONT PNR ONT	ONT QUE ONT PAC PAC	PNR NCR PNR
SDR No.	20060918003	20060822004	20060720002 20060810006 20060720001 20060915002	20060815001 20060927003 20060901009	20060919005	20060920006	20060714002 20060714003	20060808009 20060717008 20060717006 20060804010 20060811002		20060706005 20060905005 2 SDRs 20060913006 20060905004 4 SDRs	20060905002 20060712002 20060925003 PNR 20060830003 20060710002 PNR	20060915001 20060725009 20060815004
Part Condition	CRACKED	S/N	CRACKED CRACKED CRACKED SHAFT BROKEN	CRACKED CRACKED CORRODED	U/S	FAILED	BURNT FAILED	CRACKED BROKE UNSERVICEABLE CRACKED WORN TEETH	20060814003	FAILED CRACKED FAILED DAMAGED	SHEARED CRACKED BROKEN 20060816006 DRIVE SHEARED CUT 20060811001	CRACKED CRACKED BROKEN DRIVE
Part No.	2256136100	001A716E1000010	4A27122201 11030120115 4A314003 2043M12P03	2781032167 2735143003 MS3451L10SL4P	233009	V201	EM6081 3D249503	369H600151 369350542 369A7010 369D25146 369D25623	6505656 20060823002	9599060114 5531012326 9740926112 40424 555401203 NAS1581C3T11	AN626 1582102 6345703 DMQ1811A 486597 MS28775335 SMOKED	C1963 D0571 31B22111
Part Name	LATCH HOOK	AIR INTAKE ASSY.	NACELLE CHANNEL BALANCE WEIGHT SUPPORT HINGE ATTACH BRACKET MAIN FUEL PUMP STRAINER	HYDRAULIC LINE BATTERY SCOOP CANNO PLUG	ACTUATOR	FINLET	ECIRCULTION BLOWER DE-ICE TIMER	LANDING GEAR STRUT WINDOW UNILOCK ROLLER BEARING BLOWER BELT	BRUSH BLOCK PROPELLER HUB	COLD AIR UNIT BRACKET K601 RELAY MAIN WHEEL ASSY RUDDER ASSY.	BOLT TUBE ASSEMBLY RUDDER TRIM ARM CONNECTOR ELT ANTENNA RH DRIVE CABLE LH MLG OLEO SEAL CIRCUIT BREAKER PANEL	MUFFLER AIRBOX ASSY STARTER
IASC	5210	7160	5411 5523 5524 7310	1410 5720 7720	2760	5510	2121 3010	3213 5610 6220 6320 7921	6110 6114	2100 3221 3230 3246 5554 5610	3221 2720 5543 2562 2750 3213 2400	0000 2120 2435
MAKE/MODEL	DA 20 C1	DOKNIER 328 100	EMBRAEK EMB 110P1 EMB 110P1 EMB 110P1 ERJ 190 100IGW	SA227AC SA227AC SA227AC SA227AC	FORKER - WE F.28 MK0100	FBA 2C1	0069 Q069	369D 369D 369D 369D 369D 369D	LOCKHEED 188A 188A	PILATUS – SW PC 12 45 PC 12 45 PC 12 45 PC 12 45 PC 12 45	PIPER PA18 150 PA23 250 PA28 180 PA31 350 PA31 350 PA31 350 PA31 350	ROBINSON R44 R44 II R44 II

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RGN B NCR 9 PNR 9 PNR 2 PNR 1 PNR 7 NCR	4 PNR	5 PAC 2 PAC	7 PAC PAC			PAC PAC PNR PAC ONT OUE		PNR NCR PNR ONT PAC ATL			FINK	3 PAC	O VAR	2 ONT 7 PNR	6 PNR
SDR No. 20060725008 20060712009 2 SDRs 20060718003 200607002 20060905001 20060725007	20060929004	20060818006	20060905007 2 SDRs			20060818005 20060725006 20060927002 20060808005 20060927001 20060906007 2 SDRs		20060914003 20060802008 20060726006 20060712004 2 SDRs 20060918004	20060711004 20060808004 20060726005	20060817006 20060809002 20060907003	2 SURS	20060929003	20060725010	20060919002	20060821006
PART CONDITION CRACKED BENT STAKING WORN ROUGH FAILED WORN CRACKED	N/S	BROKEN U/S	CHAFED MALFUNCTION			ERODED OVERHAULED WRAPPING CRACKED CRACKED WRONG STANDARD		BROKEN BROKEN CRACKED BROKEN	BROKEN LEAKING LEAKING	DELAMINALED CRACKED FOULED	CNACNED	BROKEN	BROKEN	CRACKED FAILED	CRACKED
PART NO. C2615 NAS4283A12 B3454 C0411 8187B 1052947 C16932	50104882A	66WBL200 7615009100053	3055383			6887167 23076061 03807205 23056109 23053990 23076977 23079436		U/S CHAFED WIRE EBB124A A4871 SLC36005F LW38275	LW13521 ES48110	51203 LW121127	13828	941521	416453	30728692	4029T16P13P15
PART NAME RIB ASSY ADJUSTMENT BOLT T'R PITCH LINK DAMPER BEARING MOTOR SHAFT MUFFLER/TAILPIPE ASSY	MAIN WHEEL	BOLTS MR BLADE	WIRE, 4 GAUGE SPEED SWITCH			CASE HALVES PT GOVERNOR FUEL FILTER SCROLL SCROLL ASSY. HP1 TURBINE BLADES		AMPCONNECTOR POINTS COVER BENDIX DRIVE RETAINER MILLENIUM CYLINDER BOLT	DISTRIBUTOR GEAR CONNECTING ROD BEARING OIL FILTER	KING SE1 EXHAUST TRASITION SPARK PLUGS	CRAINICASE CASTING	SCREW	VALVE PUSHROD	SPLINE BULL GEAR	PIN, FAN BLADE RETAINING
JASC 5411 6310 6420 6510 7314 7414	3246	2910 0000	2432 7320			7230 7323 2821 7230 7230 7250		2421 7414 8011 8530 8530 7120	7414 8520 7920	8530 8120 7421	8270	6122	8530	2435 7210	7230
Make/Model R44 II R44 II R44 II R44 II R44 II R44 II	SAAB 340B SIKORSKY	S61L S76A	SWEARINGEN SA226TC SA226TC	ENGINES	ALLISON	250-C20 250-C20 250-C20B 250-C28 250-C30P AE-3007A1 AE-3007A13	AVCO LYCOMING	10-360-L2A 10-540-AE1A5 0-235-L2CM 0-320-B2C 0-320-D2J 0-360-E1A6D	O-540-F1B5 O-540-F1B5 TIO-540-A2B	110-540-CIA TIO-540-F2BD TIO-540-12BD	ROMBARDIER ROTAX	914 F3	CURTISS WRIGHT 982C9HE2	<i>CARRETT</i> TFE731-20AR-1B TPE331-10R-511C	GENERAL ELECTRIC CF34-3B1

SDR No. RGN 20060706003 PNR 20060704018 QUE 20060731002 ONT 20060731002 ONT 20060929005 PAC 20060921004 ONT 20060712005 ONT 20060704016 PAC 20060704006 PNR 20060724003 PNR 20060804004 PNR 20060906022 PAC 20060906022 PAC 20060906019 QUE 20060804001 PAC 20060804001 PAC 20060804001 PAC 20060804001 PAC 20060804001 PAC 20060802007 NCR 20060804005 NCR 20060804001 PAC 20060804002 NCR 20060906019 QUE 20060906019 QUE 20060906019 QUE 20060906019 QUE 20060906019 QUE 20060906019 <	20060704011 PAC 20060714001 ONT 20060725002 ONT 20060901004 OUE 200609117001 PNR 200609118007 ATL 2 SDRs ATL 2 SDRs PAC 2006091007 OUE 2006091007 OUE 2006091007 OUE 200609117002 ONT 20060921001 PAC 20060912001 PAC 20060912001 PAC 20060912001 PAC 20060912001 PAC 20060912001 PAC 20060912001 PAC 20060912001 PAC 20060912001 PAC 20060912001 PAC 20060916004 PAC 20060916004 PAC
SDR NO. 200607060 200607040 200607310 200607120 200608210 2006080210 200608020	20060704011 20060714001 20060725002 2006091004 20060717001 2 SDRs 20060918007 2 SDRs 2006091007 2 SDRs 2006091007 2006091007 20060921001 20060921001 20060921001 2006096012 20060916004 3 SDRs
WORN FRACTURED FRACTURED MATERIAL LOSS U/S FAILED FRACTURED FRACKED NEW FAILED FRALED FRALED FRALED FRALED FRALED FRALED FRALED FRACTURED NEW FRACTURED FRALED FRACTURED FRACTURED FRACTURED FRACTURED LEAKING U/S LEAKING U/S LEAKING SEIZED WORN U/S LEAKING SEIZED WORN U/S SEIZED WORN U/S SEIZED WORN U/S CEAKING FRACTURED DIRTY SHAFT SHEARED FRACTURED BROKEN CRACKED	FAILED LOOSE FAILED HAIRLINE CRACKED SEPARATED CRACKED WORN BROKEN SCIRED CRACKED LOOSE U/S FAILED U/S-T1 WHEEL RUB FAILED
PART NO. 310562601 3007389 310152501 310926302 3031300 32447531917 3031829 3031829 3031829 3031829 3031829 3031829 3031829 30344700 310447001 7898426009 87620136055057 MS20219A4 311496001 311717503 73030SOCN8173881	7045402L2 793651 7436024 399359CR 11876 11876 BRH20351 64936849A1 656762 649134 T1ST712ACA 05501579 955075400 70BM011030 2292152810 70BM035420 2292253850
BLED VLV INTERCONNECT ROD PRESSURE OIL ADAPTER COMPRESSOR TURBINE BLADES 1ST STAGE PLANET GEARS LARGE EXIT DUCT ENGINE FCU P3 PNEUMATIC TUBE P3 LINE IDLER GEAR ENG OUTER BURNER CAN TORQUE LIMITER POWER SECTION OIL TRANSFER ELBOW SEAL, ALTERNATOR DRIVE ELECTRONIC CONTROL UNIT SEAL TUBE ASSYS PULLEY COUPLING MECHANICAL FUEL CONTROL SEAL DIFFUSER TUBE OIL PRESSURE TRANSDUCER FUEL METERING UNIT NO.4 BEARING GASKET	FUEL CONTROL UNIT OIL TUBE FCU CYLINDER HEAD EXHAUST VALVE PUSHROD EXHAUST VALVE PUSHROD CLUSTER GEAR CRANKSHAFT CYLINDER MUFFLER HEATER IGNITOR ACCESSORY GEARBOX COMPRESSOR MO3 GAS GENERATOR 1ST STAGE TURBINE BLADE
LASC VADA 7200 7200 7210 7210 7230 7230 7230 7230 7230 7240 7250 7271 7271 7200 7320	7321 7920 7200 8530 8530 7250 7322 8011 8520 7322 8530 7421 8530 7421 8530 7230 7230
MAKE/Model IAS PRATT & WHITNEY-CANADA 720 PT6A-114 720 PT6A-21 725 PT6A-28 721 PT6A-34 731 PT6A-50 732 PT6A-60 732 PT6A-67 732 PT6A-67B 724 PT6A-67B 724 PT6A-67B 724 PW120A 724 PW120A 725 PW120A 725 PW120A 725 PW120A 725 PW123D 732 PW124 772 PW325A 725 PW33A 725 PW35A 725 PW35A 753 PW35A 753	PRATT & WHITNEY-USA JFTD12A-4A 73 JT8D-15 79 JT8D-9A 72 JT8D-9A 72 R-1340-59 85 WASP S3H1 72 BR700-715A1-30 72 TELEDYNE CONTINENTAL 85 IO-240-B 80 IO-240-B 80 IO-520-F 85 IO-520-F 85 CO-300-A 74 ARRIEL 1B 74 ARRIEL 1B 72 ARRIEL 1D1 72

Make/Model	JASC	PART NAME	Part No.	PART CONDITION	SDR No.	RGN
WILLIAMS FJ44-3A	7700	FADEC	76715	FAILED	20060901003	PNR
WSK PZL KALISZ ASZ-62IR-M18	7322	CARBURETOR	AKM621RA	LEAKING	20060925002	ONT
PROPELLER						
HAMILTON STANDARD 14SF-23 14SF-7	6120 6112	SHAFT BULKHEAD	87620130101 7849141	BROKEN SEPARATED	20060705004 20060905006	NCR
HARTZELL						
HC-B3R30-1E HC-B5MP-3C/M10876K HC-E4N-3G HD-E6C-3B PHC-G3YF-1RF	6111 6111 6110 6123 6113	BLADE BLADE THRUST PLATE, HYDRAULIC UNIT INNER SPINNER SUPPORT	R1015255 M10876ANS C459 D59901 C35325P	U/S U/S DAMAGED FAILED CRACKED	20060928005 20060926005 20060718002 20060831006 20060714007	PAC PAC PNR PNR
<i>MCCAULEY</i> 1C160/DTM7557 3AF32C D2A34C58	6110 6123 6114	FWD BULKHEAD ELECTRICAL HARNESS FERRULE	05503214 C1650130604 C3054	CRACKED WRONG PART U/S	2 SDRs 20060821002 20060928006	PAC ONT PAC
EQUIPMENT						
ABS 32100021	3246	FLANGE RETAINING RING	50140251	CRACKED	20060824002	ATL
AIR CRUISERS DIV GAR 214052200105 214052200107		LH FLOAT BAG AFT FLOAT BAG	214052200105 214052200107	GOOD GOOD	20060911004 20060911003	PAC PAC
BE AEROSPACE 72067002		OVEN	72067002	U/S	20060725003	LNO
BEECH AIRCRAFT CORP 99810047		CLEVIS	99810047	SUBSTANDARD	20060906010	PNR
BELL HELICOPTER CO. 212040001123		SUPPORT CASE	212040054007	CRACKED	20060712007	PAC
CESSNA AIRCRAFT CORP 17540091 55MM	_	EXHAUST RISER MANUAL	17540091 CHAPTER272100	CRACKED MISPRINT	20060731004 20060707004	PAC
ELECTRO-MECHANICAL D 503890571	Q'	SHAFT SHEARED		USED	20060724005	PNR
FORD MOTOR CO DOFF10300J		RECTIFIER ASSEMBLY	ES4113	DESTROYED	20060808008	PAC
JEI ELECTROMICS & LE VG204AB	3422	PRELOAD SPRING	568195301	FAILED	20060831003	PAC
MHB6018 MZ4222R		FUEL PUMP STARTER BRUSH HOLDER	200F5004R MHB6028	OVERHAULED USED BINDING	20060831001 20060824001 20060925006	ONT PNR PAC
KING RADIO CORP 071403701		FAN MOTOR	TBL25BLOWER	MELTED, BURNT	20060713002	PAC

c Pa	JASC PART NAME	Part No.	Part Condition	SDR No.	RGN
STARTER		PM1201	OVERHAULED	20060802006	ONT
PIN		101505	TIME EX	20060801002	ONT
G SWITCH FIT		DEADOUTO	WEAK	20060831005	PNR
G SWITCH		1.0400010	WEAN FAILED TEST	20060707003	PNR
ACTUATING SOLENOID			UNSERVICEABLE	20060718004	PAC
STARTER		14924HTH	INTERMITTENT	20060926002	PNR
SWITCH PLACARD		TEL82	SERVICEABLE	20060907001	ONT
STARTER		C12ST2	WORN	20060707006	ONT

AME Symposia / Trade Shows / Workshops 2007

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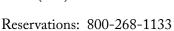


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HANGAR NOISE

Confident Your Aircraft is Airworthiness Directive compliant? How to make sure.

Introduction:

Transport Canada Civil Aviation (TCCA) has recently been advised of questions generated by some end users of our Continuing Airworthiness Web Information System (CAWIS), in particular the Airworthiness Directive (AD) module. Some users seem to think that CAWIS will provide a comprehensive list of applicable ADs on their aircraft by simply entering the aircraft registration mark. Although, CAWIS will provide a list by aircraft registration marks, it cannot be considered comprehensive.

This article will address the AD module of CAWIS and the aircraft owner's ability to retrieve applicable ADs for his aircraft. A step-by-step approach to achieving this task will be outlined below.

Functionality:

The intention of CAWIS is to provide a listing of ADs applicable to your aircraft, by registration mark, or make and model. Application and aircraft configuration plays a part in the ability to generate an all-exclusive applicability list of ADs, based solely on the aircraft registration.

TCCA will not issued ADs against amateur-built, owner-maintained or ultra-light aircraft. AD are issued against certified aeronautical products for which a type certificate (TC), supplemental type certificates (STC), limited supplemental type certificates (LSTC) and appliances certified to Technical Standard Orders (TSO), have been issued.

ADs applicable to other than the Aircraft, Engine or Propeller are identified as Miscellaneous Equipment ADs, and will not be listed against the aircraft registration mark.

Step-by-Step AD Research:

The following can be used as a guide to produce and verify a listing of ADs, which have been entered in the CAWIS database against the aircraft registration mark. This listing is in accordance with the Aircraft, Engine and Propeller Model data as identified within the database. A miscellaneous equipment AD listing will have to be produced and verified separately, regardless of aircraft registration mark entered.

First step - Airframe, Engine and Propeller ADs:

On the CAWIS AD home page, enter the last four letters of the aircraft registration mark. The list generated will provide all ADs for the Aircraft, Engine and Propeller by model applicability. It is important that the generated AD list be verified regarding Aircraft, Engine and

Propeller models stated. Aircraft information is updated upon initial registration and annually, when the Annual Airworthiness Information Report (AAIR) is completed. Therefore, the model stated may be different if it has changed since the last AAIR. This may have been done through the implementation of an STC for example. If this is the case, the "Advance Search" capabilities will have to be utilized to enter your specific product model.

A review of the AD should quickly identify if it is applicable to your particular model, serial number, configuration and/or equipment installed. After this listing has been reviewed for ADs applicable to your aircraft, retain the listing for a historical record and future reference.

Second Step - Miscellaneous ADs:

ADs applicable to anything other than the Aircraft, Engine or Propeller, are contained in a separate Miscellaneous Equipment AD list. A Miscellaneous Equipment AD listing will be produced and verified separately, regardless of aircraft registration mark, make or model.

A Miscellaneous Equipment AD list can be produced using the "Advance Search" window. Within this window use the "List Miscellaneous Equipment ADs" and click on "All ADs". This extensive listing will be alphabetical by Manufacturer or STC number regardless of the registration mark. The initial review of this listing, and determination if the ADs are applicable to your aircraft, will be a tedious task. After this listing has been reviewed for ADs applicable to your aircraft, and after all applicable ADs have been identified, retain the listing for a historical record and for future reference.

Owners of amateur-built, owner-maintained or ultralight aircraft can also use "Advance Search" function to verify ADs applicable against equipment installed on their aircraft. ADs may appear against the registration marks if your aircraft model was Type Certified (TC) and is now owner-maintained.

Third Step - Maintaining the AD applicability lists:

The airframe, engine and propeller AD list generated within step one, can be easily maintained by using the "Advance Search" window. Locate the statement "Find Specific ADs by...Registration Mark C-". Enter the last four letters of your aircraft registration mark and click on "Recent ADs". The generated list (aircraft, engine and propeller models) will be recent ADs according to aircraft registration mark.

Within the "Recent ADs" list, ADs added to CAWIS within the past seven days (New) will be identified by a check mark. Those ADs without a check mark have been added within the last thirty days. If no ADs have been added within the last thirty days the generated list will be blank under each of the Aircraft, Engine and Propeller model titles.

The Miscellaneous Equipment AD list generated within step two, can also be easily updated by using the "Advance Search" window. Locate the statement "List Miscellaneous Equipment ADs" and click on "Recent ADs". This list will be all Miscellaneous Equipment ADs added to CAWIS regardless of aircraft registration mark.

Miscellaneous Equipment ADs added to CAWIS in the past month "New" will be identified by a check mark. Those ADs without a check mark have been added within the last six months. If no Miscellaneous ADs have been added within the last six months the generated list will be blank.

On the AD home page there is a "Recent" function that can be used. The list generated is all ADs added to CAWIS regardless of aircraft registration mark or aircraft manufacture. This list will also have all miscellaneous equipment ADs. The same criteria as identified above regarding New or later added ADs will apply.

Summary:

The three steps indicated above should be helpful in review, verification and determination of applicable ADs against your aircraft model in accordance to serial number, configuration and equipment installed.

Read the fine print on the screen to be alerted of the functionality and cautions to be aware of during your AD review. Take note of the fine print on the web pages and review the *Canadian Aviation Regulation 593* for aircraft owner responsibilities to request continuing airworthiness information directly from the manufacturer.

Aircraft owners should establish a scheduled time to verify compliance with all ADs applicable to their products and when uncertain of the applicability against your aircraft, contact the nearest Transport Canada Civil Aviation Regional Office.



Brad Ford

Pacific AME Symposium in Vancouver

FAA UNAPPROVED PARTS NOTIFICATION (UPNs)

Published by: FAA, AIR-140, P.O. Box 26460, Oklahoma City, OK 73125. UPNs are posted on the Internet at: http://www.faa.gov/avr/sups/upn.cfm

NO. 2006-00058

ISSUED 5 OCTOBER 2006

Affected Parts

Hot air balloons.

Purpose

The purpose of this notification is to advise all aircraft owners, manufacturers, maintenance organizations, and parts suppliers and distributors regarding improper maintenance performed on hot air balloons.

Background

Information received during a Federal Aviation Administration (FAA) suspected unapproved parts investigation revealed that between August 2002 and November 2005, Micki's Balloon Repair (Micki's), located at 4005 W. Pinecrest Drive, Marshall, TX 75670, improperly repaired and approved for return to service various hot air balloons. Micki's holds FAA Air Agency Certificate No. MKOR497X.

Evidence indicates that Micki's approved for return to service hot air balloons that were not maintained in accordance with the methods, techniques, and practices prescribed in the current manufacturer's maintenance manual or Instructions for Continued Airworthiness.

Discrepancies noted in Micki's practices included, but are not limited to, the following:

- The use of unapproved fabric in the repair of various balloons. No certificate of equivalency could be produced for the fabric used. Specifically, Micki's used a Kenyon Industries, Inc., 70D/34 1.9 OZ RIP T95 T66 fabric, which is not authorized by the Aerostar maintenance manual.
- Failure to produce certification documents for various rolls of balloon material used in the repair process.
- Failure to maintain a current repair station/ quality manual.
- Failure to maintain tools that are used to make airworthiness determinations in a current calibrated status.
- Failure to properly segregate unserviceable parts from serviceable parts.

Recommendations

Regulations require that type-certificated products conform to their type design. Aircraft owners, operators, maintenance organizations, and parts distributors should inspect their aircraft, aircraft records, and or parts inventories for any balloon repair work accomplished by Micki's between August 2002 and November 2005. If any repairs were performed, appropriate action should be taken. If any of the referenced fabric is found in existing inventory, it is recommended that the fabric be quarantined to prevent installation until a determination can be made regarding the fabric's eligibility for installation.

Further Information

Further information concerning this investigation, and guidance regarding the above-referenced maintenance, can be obtained from the FAA Flight Standards District Office (FSDO) given below. The FAA would appreciate any information concerning the discovery of the fabric from any source, the means used to identify the source, and the action taken to remove the fabric from aircraft and/or stock.

This notice originated from the FAA Dallas FSDO, 3300 Love Field Dr., Dallas, TX 75235, telephone (214) 902–1800, fax (214) 902–1862; and was published through the FAA Suspected Unapproved Parts Program Office, telephone (703) 668–3720, fax (703) 481–3002.

NO. 2002-00062

ISSUED 3 NOVEMBER 2006

Affected Parts

Various aircraft parts.

Purpose

The purpose of this notification is to advise all aircraft owners, operators, manufacturers, maintenance organizations, and parts distributors regarding the unapproved status of aircraft parts sold by Standby Parts, Inc., an aircraft parts distributor.

Background

Information received during joint investigations conducted by the Federal Aviation Administration (FAA), the Department of Transportation – Office of Inspector General, and the Federal Bureau of Investigation revealed that between September 1999 and April 2002, Standby Parts, Inc. (Standby Parts), a parts distributor previously located at 120 Penn Street, El Segundo, CA 90245, sold aircraft parts using fraudulently produced documents.

Evidence indicated that Standby Parts made or caused to be made invoices, part certifications, and FAA 8130-3 Airworthiness Tags that contained false statements regarding the actual condition of aircraft parts. Between

September 1999 and April 2002, Standby Parts purchased various aircraft parts known to be in a "repairable" or "as is" condition. Through Daniel Larue Booker, an FAA Designated Airworthiness Representative (DAR), and George G. Thompson, owner of TATCO, a repair station, Standby Parts obtained FAA 8130-3 tags that falsely represented the parts as being in a new condition. Some of the parts inspected and approved for return to service by TATCO were outside the repair station's ratings and limitations. Some of the FAA 8130-3 tags issued by Daniel Booker were for parts never inspected by the DAR.

Recommendations

Regulations require that type-certificated products conform to their type design and be properly maintained. Aircraft owners, operators, manufacturers, maintenance organizations, and parts distributors should inspect their aircraft and/or parts inventory for aircraft parts sold by Standby Parts between September 1999 and April 2002. If any are found installed on aircraft, appropriate action should be taken. If any are found in existing aircraft stock, it is recommended that they be quarantined to prevent installation until a determination can be made regarding their eligibility for installation.

Further Information

Further information concerning this investigation, and guidance regarding the above-referenced parts, can be obtained from the FAA Flight Standards District Office (FSDO) shown below. The FAA would appreciate any information regarding the discovery of the above-referenced parts from any source, the means used to identify the source, and the actions taken to remove the parts from aircraft and/or stock.

This notice originated from the FAA Los Angeles FSDO, 2250 East Imperial Highway, Suite 140, El Segundo, CA 90245, telephone (310) 215–2150, FAX (310) 649–5680; and was published through the FAA Suspected Unapproved Parts Program Office, AVS–20, telephone (703) 668–3720, FAX (703) 481–3002.

NO. 2006-00157

ISSUED 15 NOVEMBER 2006

Affected Parts

Aircraft accessories and instruments.

Purpose

The purpose of this notification is to advise all aircraft owners, operators, manufacturers, maintenance organizations, and parts suppliers and distributors regarding improper maintenance performed on aircraft accessories and instruments.

Background

Information received during a Federal Aviation Administration (FAA) suspected unapproved parts investigation revealed that Fat Angel Aviation Services, Inc. (Fat Angel), located at 737 South Point Blvd. Suite G, Petaluma, CA 94954, improperly maintained and approved for return to service aircraft accessories and instruments. Fat Angel formerly held Air Agency Certificate No. OFGR270L with Accessory Class 1 and 2 ratings, and Limited rating (specialized services). The FAA has not been able to determine the total number of parts affected or the timeframe in which the improper maintenance occurred. Discrepancies noted in Fat Angel practices included, but are not limited to, the following:

- Maintaining and approving for return to service various instruments without holding an instrument rating.
- Approving for return to service instruments and accessories described as having been repaired without being inspected or repaired using acceptable methods, techniques, and practices.
- Performing instrument and accessory repairs without using required approved data.
- Failing to maintain instruments and accessories in accordance with Continuous Airworthiness Maintenance Program manuals.
- Failing to properly document instrument and accessory repairs.

Recommendations

Regulations require that type-certificated products conform to their type design. Aircraft owners, operators, manufacturers, maintenance organizations, and parts suppliers and distributors should inspect their aircraft, aircraft records, and/or parts inventories for any aircraft instruments or accessories that were approved for return to service by Fat Angel.

If these instruments or accessories are found installed on aircraft, appropriate action should be taken.

If instruments or accessories are found in existing inventory, it is recommended that they be segregated to prevent installation until their eligibility for installation is determined.

A partial list of instruments and accessories that have been approved for return to service by Fat Angel can be viewed at http://www.faa.gov/aircraft/safety/programs/sups/upn/media/2006/UPN 2006-00157 Partial Parts List.doc.

Further Information

Further information concerning this investigation, and guidance regarding the above-referenced instruments and accessories, can be obtained from the FAA Flight Standards District Office (FSDO) given below. The FAA would appreciate any information concerning the discovery of the above-referenced instruments and accessories from any source, the means used to identify the source, and the actions taken to remove the instruments or accessories from aircraft and/or stock.

This notice originated from the FAA Oakland FSDO, 1420 Harbor Bay Parkway, Suite 280, Alameda, CA 94502, telephone (510) 748–0122, fax (510) 748–9559; and was published through the FAA Suspected Unapproved Parts Program Office, AVS–20, telephone (703) 668–3720, fax (703) 481–3002.











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CIVIL AVIATION INTERNET SITES:

Aviation Information

www.tc.gc.ca/civilaviation/menu.htm

Canadian Aviation Regulations (CARs)

www.tc.gc.ca/civilaviation/regserv/affairs/cars/menu.htm

Airworthiness Directives

www.tc.gc.ca/CivilAviation/certification/continuing/ad.htm

Service Difficulty Alerts

www.tc.gc.ca/CivilAviation/certification/continuing/Alert/menu.htm

Service Difficulty Advisories

www.tc.gc.ca/CivilAviation/certification/continuing/Advisory/menu.

Web Service Difficulty Reporting System (WSDRS)

www.tc.gc.ca/wsdrs/

Airworthiness Notices

www.tc.gc.ca/civilaviation/maintenance/aarpc/ans/menu.htm

Airworthiness Manual Advisory Index

http://www.tc.gc.ca/CivilAviation/certification/guidance/menu.htm

Aircraft Maintenance & Manufacturing Staff Instructions (MSI)

www.tc.gc.ca/civilaviation/maintenance/aarpc/msi/menu.htm

Aircraft Maintenance and Manufacturing Policy Letters (MPL)

www.tc.gc.ca/civilaviation/maintenance/aarpc/mpl/menu.htm